ROTORCRAFT FLIGHT SIMULATION WITH AEROELASTIC ROTOR AND IMPROVED AERODYNAMIC REPRESENTATIONS. VOLUME III. PROGRAMMER'S MANUAL

P. Y. Hsieh, et al

Bell Helicopter Company

Prepared for:

Army Air Mobility Research and Development Laboratory

June 1974

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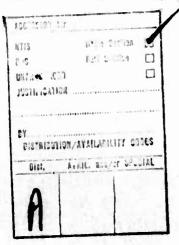
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This report has been reviewed by the Eustis Directorate, U. S. Army Air Mobility Research and Development Laboratory and is considered to be technically sound.

The computer program resulting from this contract will be provided, upon request of qualified users, for use in the design and analysis of rotary-wing aircraft. Volume III of this report, a programmer's manual, has not been widely distributed, but will be provided with the computer program to aid in program installation.

The technical monitor for this contract was Mr. Edward E. Austin, Aeromechanics, Technology Applications Division.



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18. KEY WORDS (Continue on reverse side if necessary and identify by block number)

Aerodynamics

Flight Simulation

Dynamics

Aeroelasticity

Helicopters

Structural Properties

Rotors

Control

Computer Programs

Numerical Analysis

Digital Computars

Motary Wing Aircraft

20. ABSTRACT (Centinus et : see to olds if necessary and identity by block number)

This report consists of three volumes and documents the current version in the C81 family of rotercraft flight simulation programs developed by Bell Helicopter Company. This current version of the digital computer program is referred to as AGAJ75x

The new, revised, or alternate mathematical models incorporated into the program during the current contract are as follows:

D 1000 1473 EDITION OF 1 NOV 65 IS OBSOLETE

Inclassified

# Block 20. Continued

- (1) Fuselage aerodynamic forces and moments (revised)
- (2) Aerodynamic surfaces (revised with two surfaces added)
- (3) External stores/aerodynamic brakes (new)
- (4) Rotor blade airfoil section distribution (new)
- (5) Rotor-induced velocity distribution (alternate)
- (6) Rotor unsteady aerodynamics (alternate)
- (7) Rotor wake effect at aerodynamic surfaces (alternate)
- (8) Method for numerically integrating rotorcraft equations of motion (alternate)

This volume, the Programmer's Manual, contains the information necessary to set up and support the computer program. Specifically, it includes cross-references of FORTRAN COMMON BLOCK variables, a catalog of subroutines, and a discussion of programming considerations. The listings and related soft-ware for the computer program documented in this report are unpublished data which are on file at the Eustis Directorate, U. S. Army Air Mobility Research and Development Laboratory (USAAMRDL), Fort Eustis, Virginia. Volume I, The Engineer's Manual, documents the background and development of the current version of the program. Volume II, the User's Manual, contains the detailed information necessary for setting up an input data deck and interpreting the computed data.

# PREFACE

This report and its accompanying computer program were developed under Contract DAAJ02·72-C-0098 awarded in June 1972 by the Eustis Directorate of the U. S. Army Air Mobility Research and Development Laboratory (USAAMRDL). In addition to the work performed under this contract, the report and computer program include the documentation and program features developed under USAAMRDL Contracts DAAJ02-70-C-0063 and DAAJ02-73-C-0086. The contractor and USAAMRDL have agreed that the computer program documented herein is the new master version of the program. Hence, this report supercedes all previous versions of the C81 program and documentation.

Technical program direction was provided by Mr. E. E. Austin of USAAMRDL. Principal Bell Helicopter personnel associated with the current contract were Messrs. B. L. Blankenship, J. M. Davis, and P. Y. Hsieh, and Dr. B. T. Waak. In addition, Dr. R. L. Bennett and Mr. B. J. Bird assisted in coordinating the work and documentation prepared under the two previous contracts noted above with that prepared under this contract.

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# 1. INTRODUCTION

This manual documents the rotorcraft flight simulation program, designated AGAJ73, and its post processor for data reduction, designated GDAJ07. To the user, this system appears as a single program; to the programmer, the two programs are very different. This documentation is for the programs as they were written for, and are being used on, an IBM System/360 Model 65 Computer at Bell Helicopter Company.

The information in this volume is of two types. Section 2 contains the information necessary to get the programs operational on a computer compatible with the statellation at Bell Helicopter Company. If the programs are to be modified in any way, the programmer will need the information in Sections 3 and 4 of this volume.

# 2. OPERATING ENVIRONMENT AND PROCEDURES

The System/360 environment under which this program system is maintained is Operating System/360 Option IV (MVT). Input on the system reader is controlled by Houston Automatic Spooling Program (HASP) II, as is system output destined for on-line printer or card punch. As delivered under this contract, the program requires approximately a 1000K region of memory (1K = 1024 bytes, 1000K ~ 1 megabyte). Methods for reducing this large memory requirement are discussed below and in Section 4.4. The Basic Sequential Access Method routines are retained in a resident LINKPACK area. Scratch data sets are directed to a Telex 5312 Direct Access Storage Facility. Tape data sets are recorded on Telex 6420 Tape Drives. The recording format for the CALCOMP PLOTTAPE is 9 track, 800 bpi, NRZI.

All other tape data sets are recorded as 9 track, 1600 fci phase encoded. To run the program under OS/360, at least 262,144 (256K) bytes of main memory are required for operation with Option I (PCP) or Option II (MFT) systems. At least 524,288 (512K) bytes of main memory are required for Option IV (MVT).

The program has been maintained on the IBM System/360 FORTRAN H Compiler in USA FORTRAN. Compiler options used are SOURCE, EBCDIC, NOLIST, NODECK, LOAD, MAP, NOEDIT, ID, XREF, and OPT=2. Since the FORTRAN H Compiler performs essential optimization functions, compilation of this program under any other compiler or using optimization level less than "2" on the H Compiler will result in degraded performance in speed. Some FORTRAN language features peculiar to IBM FORTRAN definitions have been noted explicitly in individual routines.

AGAJ73 can be link-edited in several different ways. For example, with the OVERLAY structure shown in Table 2-1, the entire program can be loaded into either main or extended memory; with the HIARCHY support structure shown in Table 2-2, part of the program can be loaded into main memory and the rest into extended memory; or if neither OVERLAY nor HIARCHY structure is used, the entire program can be loaded into either main or extended memories. The best way to link-edit this program depends upon the facilities available at the local installation.

Although this program is quite large, it can be run in less than 300K if the OVERLAY structure shown in Table 2-1 is used and the size of several large arrays is reduced according to the procedure outlined in Section 4.4. If the region size available is greater than 300K, the OVERLAY shown in Table 2-1 can be relaxed to reduce the overhead time associated with OVERLAY. By using the HIARCHY support structure shown in Table 2-2 and also following the procedures in Section 4.4, the program can run in a 206K region of main memory and a 450K region of extended memory. It would need approximately another 300K if the reduction of array sizes mentioned above is not done. Since the buffers are in extended memory, the amount of extended memory used depends upon the units opened.

The input data for the linkage editor for GDAJ07 are shown in Table 2-3. This OVERLAY structure results in a program whose longest segment is 142K bytes. After a buffer allocation, the size of the region needed is 184K bytes.

The Job Control Linkage (JCL) used to run a typical set of data is shown in Figure 2-1. If other than HIARCHY structure is used, the REGION card in the first step should be changed accordingly from that shown in Figure 2-4. Tables 2-4 and 2-5 give the unit type, what it is used for, and which subroutines refer to it for AGAJ73 and GDAJ07, respectively.

The JCL has been prepared such that the job is in two steps: C81 and C81PLOT. The C81 step executes AGAJ73, and the C81PLOT step executes GDAJ07 even if the first step abends. This was done so that if a maneuver abends on the first step, the user will still get plots of what was run.

# TABLE 2-1. LINKAGE EDITOR CARDS FOR OVERLAY OF AGAJ73

```
OVERLAY ALPHA INITIALIZATION SEGMENT
INSERT
        PDSRED, REDID, REDRWK, REDSWK, START, WKTABN, IHCNAMEL
OVERLAY BETA
INSERT
         ERRCHK, JSTRED, LGCINT, LIZE, NPUTOT, RBDY, READIN
INSERT
        REDATB, REDBMS, REDCL, XSTINT, ZERO
OVERLAY BETA
INSERT
         FUSINT.INBLD.INPU.INRTR.INSCAS.JFBGIN.MODAL.PYLINT
INSERT
        RTINIT, STBZIN, WPMDAL, XCONIN
OVERLAY BETA
INSERT
        MNEM, TABFIX, TABOUT, TURN, YRINIT, YSINIT
OVERLAY ALPHA HARMONIC ANALYSIS
        HARM, LOADT
INSERT
OVERLAY ALPHA
                STABILITY ANALYSIS SEGMENT
INSERT
         ALLMAT, ALSTAB, INVERS, IDMAT, MORDRS, MODES, PUNCH, WRMS
INSERT
         IHCCLABS, IHCCLAS, IHCCLSQT, IHCCSABS, IHCGSAS, IHCFOPT
INSERT
         IHCLSORT
OVERLAY ALPHA GENERAL PURPOSE SEGMENT
INSERT
         ANAL, ANDOIT, AZMINT, AZMUTH, CDCL, CLCD, CMCALC, DIFFER
INSERT
         DOTX, FOCUS, FORCMC, FUSENM, HRESP, INTERQ, ITROT, MBAL
INSERT
        NSTED . PADBGN, RADIAL, RADOUT, RGUST, ROTAN, RTWAKE, SOLVE
INSERT
         STBWAK, STBZFM, SWSRAT, TABINT, TRMANU, UNSTED, WING, WRFM
         WRTMNV, WSHDUF, XSTORE, IHCSASCN
INSERT
OVERLAY BETA
INSERT
        AJACOB, JACOBI, DAMPER, WRVP
                STABILITY ANALYSIS SEGMENT
OVERLAY GAMMA
INSERT
        WRINST
OVERLAY DELTA
INSFRT
         INSTAB
OVERLAY DELTA
INSERT
        STAB, WRSTAG
OVERLAY GAMMA TRIM SEGMENT
INSERT
        TRIM, TVT
OVERLAY DELTA
INSERT
         RPTPG
OVERLAY DELTA .
INSERT
        TRM1
OVERLAY DELTA
INSERT
        WRTRIM
OVERLAY DELTA
```

INSFRT

ITRIM, PDZ1

#### TABLE 2-1. Concluded.

MANEUVER SEGMENT OVERLAY BETA INSERT DER IV. HAMMS . HPCG. MANU. MANUV, QUAN, RNGKTA, SAVE IC. WAG OVERLAY GAMMA INSERT SCASIT OVERLAY GAMMA INSERT VARI OVERLAY DELTA INSERT VCNTRL OVERLAY DELTA INSERT SUPERP OVERLAY DELTA INSERT MTLT OVERLAY DELTA INSERT GUST OVERLAY DELTA INSFRT **FXTORS** TYPRLAY DELTA INSERT CNTM OVERLAY DELTA INSERT MOMB OVERLAY ONE (REGION) INSERT INIT, RESTRT, SIVAR, TIMLP, TIVAR, WRMANU OVERLAY ONE INSERT CONSTR.STBD ENTRY MAIN

TABLE 2-2. LINKAGE EDITOR DATA CARDS FOR HIARCHY SUPPORT OF AGAJ73

	l			
HEARCHY 1. II			HEARCHY S	1,SCASIT
HIARCHY 1, II	HEECOMH HIARCHY			L.SIVAR
HEARCHY 1. II		1. TOMAT		L.STAMAN
HEARCHY 1. E		The second secon	HEARCHY !	
HIARCHY 1. IF		1,JFBGIN		1.START
HIARCHY 1.11	HCETRCH HIARCHY		HI ARCHY	
HIARCHY 1. II	HCFC VTH HE ARCHY		HI ARCHY	
HEARCHY 1. II			HEARCHY	
HEARCHY 1. II	HCNAMEL HI ARCHY			1.SWSRAT
	HCUATBL HIARCHY			1.TAOFIX
HIARCHY 1.II	HCUOPT HEARCHY			1,TABOUT
	BOUMP HEARCHY		HI ARCHY	_ •
HEARCHY 1.A.	JACOS HEARCHY	1, MORDAS	HE ARCHY	
HIARCHY 1.AL	LSTAB HEARCHY			1,TIVAR
HIARCHY 1.CO	BZARM HEARCHY			1, TOPLOT
HEARCHY 1.C	NTH HEARCHY			1.TRIM
HIARCHY 1,CC	DNSTB HEARCHY			1, TRHANU
	ONTRH NEARCHY		HIARCHY	
HIARCHY 1.DA	ATE HEARCHY	1.NOPS		1.TURN
	AMPER HEARCHY			1, VARI
HIARCHY 1.DE	ERTY HEARCHY			1. VENTRL
HIARCHY 1.EF	rrchk htarchy			1,WAG
HIARCHY 1.E	KTORS HEARCHY	1.PUNCH	HI ARCHY	
HIARCHY 1.FC	OCUS HE ARCHY	1. PYLINT	HE ARCHY	
HIARCHY 1.FC		1.OUAN		1. WRINST
	DRUK1 HEARCHY		HEARCHY	1. WRMANU
	DRYYY HIARCHY			I, WAMDAL
The state of the s		1,READ:N	HEARCHY !	
HIARCHY 1.FO		1.REDATA		1, WROT1
HIARCHY 1.FO			HI ARCHY	
	USINT HIARCHY			1.WRTHWV
HEARCHY 1.6		1,REDID		1. WRTRIM
HIARCHY 1. H		1.REDRWK		1. HRVP
HIARCHY 1.HF			HEARCHY !	
HIARCHY 1. IN				1,XSTINT
HEARCHY 1. IN				1. YRINIT
HEARCHY 1. IN			HEARCHY	
HI ARCHY 1. IN			HI ARCHY	
	ISCAS HEARCHY		HI ARCHY	
HIARCHY 1, IN	ISTAB HEARCHY	1.SAVOĻO	ENTRY MA	IN

# TABLE 2-3. LINKAGE EDITOR DATA CARDS FOR OVERLAY OF GDAJO7

IEW0000 OVERLAY ALPHA **IFW0000** INSERT CURVET 1 FW0000 INSERT INCLATES INSERT IHCLSORT IFW0000 I EWOOOO INSERT COLL INSERT TIMPTS **IEW0000** OVFRLAY ALPHA 1 EW0000 I EWOOOO INSERT CALCOM 1EM0000 INSERT IHCFRXPI IEWOOOD INSFRT LINE IEW0000 INSERT NEXTTIME INSERT NUMBER IFW0000 INSERT SYMBOL 1FW2222 I EWOOOC OVERLAY BETA 1EW0000 INSERT FSFT **IEW0000** INSERT HARM INSERT IHCSSORT I EWODOO INSERT PLOTER IEW0000 **IEWOOOD** INSERT AXIS# INSERT **IEW0000** IHCFRXPR I EWOODO INSERT IHCLSCN INSERT IHCSEXP **IEM0000** I EWOOOD INSERT IHCSLOG 1EM0000 INSERT SCALE# OVFRLAY BETA I EMOCOO INSERT SCALIT 1EW2220 INSERT CALCUL 1EM0000 INSERT PPLOT 1 FW0000 1 FW0000 INSFRT SCLFIX IEW0000 INSERT INPLOT ENTRY MAIN IEMODDO

DENNITE

Unit No.	Туре	Used for	Used by Subroutine
1	direct access	Permanent data storage in partitioned data set (PDS)	JSTRED, REDATB, REDBMS, REDCL, REDID, REDRWK, REDSWK
2	tape	New restart tape	RESTRT
3	direct access	Utility storage of time- variant trim flapping history and storing maneuver time history to pass to GDAJ07	INIT, LOADT, MAIN, MANU, MANUV, RESTRT, TVT
4	tape	Old restart tape	RESTRT
5	card reader	Input data	MAIN
6	printer	Printed output	ALSTAB, AZMUTH, CDCL, CLCD, DAMPER, ERRCHK, EXTORS, FUSINT, HAMMS, HPCG, INRO, INSTAB, ICMAT, ITRIM, ITROT, JFBGIN, LIZE, LOADT, MAIN, MBAL, MNEM, NPUTOT, RADOUT, READIN, REDID, REDRWK, RPTPG SIVAR, STAB, TABOUT, TIVAR, TRIM, TRM1, TURN, VIND, WAG, WRFM, WRINST, WRMANU, WRMDAL WRMS, WROT1, WRSTAB, WRTMNV, WRTRIM, WRVP, XCONIN, YRINIT YSINIT
7	card punch	Punched output	PUNCH
10	direct .	Utility storage of AGAJ73 input data	JSTRED, MAIN, READIN, REDATB REDBMS, REDCL, REDID, REDRWK REDSWK
11	direct access	Passing input data to GDAJ07	MAIN, READIN

	Vacantaria (	TABLE 2-4. Continue	ed.
Unit No.	Туре	Used for	Used by Subroutine
14	direct access	Stored and retrieved initial conditions when Hamming's method is used	SAVEIC

TABLE 2-5. INPUT/OUTPUT UNITS USED IN GDAJ07						
Unit No.	Туре	Used for	Used by Subroutines			
3	direct access	Maneuver time history from AGAJ73 or Tape 8	CURVET, C81L, FSFT, MAIN, SCALIT			
6	printer	Printed output	CALC81, CURVET, C18L, FSFT, PPLOT, WROT1			
8	tape	Old time history tape	C81L			
9	tape	New time history tape	C81L			
10	direct access	Input data from AGAJ73	CURVET, C81L, FSFT, MAIN, SCALIT			
PLOT- TAPE	tape	Plot maneuver time history in GDAJ07	PLOTER			

```
//22712370 JOS (ASAJ7300.638.69109304.0P20.T.02).'PY
// MSGLEVFL-1.GLASS-C
                                                                                            2841 ..
//COLPAGE PAGE PAGG-AGAJ7306.LIB-ENGTEST.F35P-100.8LK-7292.
                                                                                                     *00000010
                    REST!=NULLFILE.RESTO=NULLFILE.TH!N=NULLFILE.
THRIIT=NULLFILF.THSFR=0.SYSPLOT=NULLFILE.
GRAPH=GDAJO7-LIB1=ENGTEST
11
                                                                                                     -00000000
                                                                                                      00000040
110
                                                                                                      00000050
//•
                    PARAMETERS ON THE EXEC STATEMENTS USAGE
                                                                                                      20000040
                                                                                                      00000070
                                                                                                      00000000
110
                                                 PROGRAM NAME
                                                                                                      00000000
             2206
                            AGA173
                            PHOR . PROD 1
                                                 LIRRARY WHERE PROGRAM RESIDES
                                                                                                      00000100
             LIB
                                                 NO. OF CYLINDERS FOR FT03F001
BLOCKSIZE OF FT03F001. FT08F001. AND
              FISP
                            100
                                                                                                      20000110
             BLE
                                                  FT09F001
                                                                                                      00000130
                                                 OSMAME FOR RESTART TAPE IMPUT
OSMAME FOR RESTART TAPE OUTPUT
OSMAME FOR TIME HISTORY IMPUT
OSMAME FOR TIME HISTORY OUTPUT
VOL-SER FOR TIME HISTORY EMPUT TAPE
OSMAME FOR FTOSPOOL
             PEST
                            NULLFILF
                                                                                                      00000140
//•
             RESTO
                            MULLFILE
                                                                                                      00000150
             THIN
//:
              THOUT
                            MILLPILE
                                                                                                      00000170
              THIP
                                                                                                      00000180
                            (TEMP)
110
             SYSUTI
                                                 ANY TAPE DRIVE
ANY DUAL DENSITY TAPE DRIVE
DSMAME FOR PLOT TAPE
110
              TPS
                                                                                                      00000200
                                                                                                      00000210
                                                                                                      000 30 220
                            MULLFILE
//0
             SYSPLOT
                                                                                                      00000230
110
             EXPC PG4-SPROG.REGION-(204K.550K)
                                                                                                      00000240
                    DISP-SHR.DSM-6LIB
UNIT-2314.DSM-68LIB72(DUM).VOL-SER-JOBLBH.DISP-SHR.
//STPPLIB
                                                                                                      00000250
//FT017001 00
                                                                                                     00000260
                    OCR-HIARCHY-1.LAREL=(...EN)
OCR-(RFCFH-FR.LRFCL-80.RLKS12E-800)
UNIT-(TPS..RFER).OISP-(.CATL3.DELETE).OSH-ERESTO.
                                                                                                      00000270
//FT02F001 00
                                                                                                     •00000290
                                                                                                      00000300
                    unit-sysna, space-(cyl, (cp3sp)).osm-esysuti.
och-(recpm-vb3.lrecl=3644.blksize-eblk,miarcmy-l).
                                                                                                     +00000310
//FT03F001 DO
                                                                                                     +00000320
                     DISPOINE . PASSI
                                                                                                     00000330
                    UNIT-ITPS..OEPPR).DISP-OLD.DSN-LRESTI.
DCB-HIAKCHY-1
//FT04F001 DD
                                                                                                      20000350
//FT05F001 D0
                    DOMAMF- IN
                     SYSOUTOA
                                                                                                      00000370
                                                                                                      00000300
                    SYSOUTOR
//FTOTFOOL DO
                                                                                                     -00000390
//FT10F001 00
                    UNIT-SYGDA. SPACE-(TRK. (10)) .DSN-65YSIN1.
                     OCH- (RECHAPS.LRECL-80.BLKSIZE-3600.HIARCHY-1).
                                                                                                     -00000400
                     DISPOINTH. PASSA
                                                                                                      00000410
                    DSM-CSYSINZ.UNIT-SYSDA. SPACE-(TRK.(10)).
//FT11F001 00
                    DCA=(RECPM-F. BLKS [ZE=80.MEARCHY=1).DISP=(MEM.PASS)
UNIT-SYSDA, SPACE=(TRK.(8.4)).DISP=(MEM.DELETE).
DCM=HIARCHY=1
                                                                                                      00000430
//FT14F001 DO
                                                                                                     *D0000440
                                                                                                      00000450
//CAIPLOT EXFC PGM-CGRAPH.COMD-114.LE.CBIJ.EVEN)
                                                                                                      00000440
//STEPLIB 00
//FT03F001 DO
                    DISPOSHED DELETED OSMO ESYSUTIODES HIARCHYOL
                                                                                                      00030470
                                                                                                      00030480
                    SYSQUI-A
UNIT-17PS..OFFFR:.DISP-OLD.DSN-ETHIN.VOL-SER-ETHSER.
DCR-HIARGHY-1
                                                                                                     000004 90
•00000 500
//FTOGFOOL DO
                                                                                                      00000510
//FT09F001 00 UNIT-(175..O.FER).015P-(.KEEP.OFLETE).05N-4THOUT.

OCG-(RECFM-VBS.LRECL-3644.BLK512E-68LK.HIARCHY-1)

//FT10F001 00 DISP-(OLO.OFLETE).05N-45Y5IN3.DG8-HIARCHY-1
                                                                                                     +00000520
                                                                                                      00000530
                                                                                                      00000540
                                                                                                     00000550
00000540
00000570
//PLOTTAPF DD UNIT-(DDT..DFPER).DSN-ESYSPLGT.LABFL-(.NL).VOL-PRIVATE
            OCA-HEAGHY-L
PEND THIS IS THE END OF THE EN-STREAM PROCEDURE
FAPE COLPAGE
"
//IN DO .
                                                                                         Reproduced from best available copy.
```

\*REGION size shown in for use with HIARCHY support. If OVERLAY is used or entire program is run only in main or only in extended memory, the REGION size must be changed accordingly.

Figure 2-1. Job Control Language To Run AGAJ73 and GDAJ07 Data.

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# 3. GENERAL PROGRAMMING AIDS

# 3.1 MARCO FLOW CHARTS

The flow charts in Figures 3-1 through 3-6 describe the functional structure of the program without regard to flow by subroutine. Figure 3-1 shows the total program structure and is a composite of Figures 1-1 through 1-13 in Volume II. Figure 3-2 provides some detail of the trim process in Figure 3-1. Figure 3-3 amplifies the stability analysis shown in Figure 3-1. Figures 3-4 and 3-5 give some flow logics of Runga-Kutta Method and hamming's Predictor-Corrector Method employed, respectively, in maneuver functions in Figure 3-1. Figure 3-6 details the data reduction program.

# 3.2 FORTRAN SUBROUTINES IN AGAJ73

The FORTRAN subroutines contained in AGAJTO are listed in alphabetical order including the main program, which is called MAIN. A few remarks are made for each subroutine which indicate its general purpose or use in the program. In the case of subroutines with multiple entry points, the names of all the entries are given after the name of the subroutine in the order that they occur in the subroutine.

- (1) AJACOB, AJACB1. This subroutine handles computation of quantities which depend upon variables which are changed in either subroutine TRIM or subroutine STAB in order to compute partial derivatives. These quantities are then calculated and used in the computation of forces and moments.
- (2) <u>ALLMAT, ALLVEC.</u> The stability analysis uses this subroutine to compute eigenvalues and eigenvectors.
- (3) ALSTAB. This subroutine calls ALLMAT and processes and prints out the results of the stability analysis.
- (4) ANAL. Output of this subroutine consists of the total summation of forces and moments.
- (5) AZMINT. It initializes some variables which will be used by subroutine AZMUTH.
- (6) AZMUTH. This subroutine in the rotor analysis does the calculation and integration at each blade azimuth position.
- (7) GDCL. This subroutine uses the local angle of attack and Mach number plus the airfoil aerodynamic inputs to compute the steady-state lift, drag, and pitching moment coefficients for each blade segment of the rotor.
- (8) <u>CGZARM. CGYARM. CGXARM.</u> This subroutine calculates moment arms about the cg for the aerodynamic surfaces whenever cg is shifted.

- (9) <u>CLCD</u>. This subroutine is similar to subroutine CDCL except that CLCD computes the three aerodynamic coefficients for the wing and stabilizing surfaces.
- (10) <u>CMC/LC</u>. This subrouting interpolates on the Carta tables to produce the contribution of unsteady aerodynamics to the pitching moment. It is the major section of the BUNS unsteady aerodynamic model.
- (11) <u>CNTM. AUXJ. FLAP</u>. Some of the forcing functions in a maneuver may be timed to start after the rotors have been stopped. This subroutine converts those relative times to absolute times. It also varies jet thrust and RPM dependent flapping stops during a maneuver.
- (12) CONSTB. This is the control program for the stability analysis.
- (13) CONTRM. This is the control program of the trim segment.
- of this is to gradually dampen out oscillations of the trim iterations. This is accomplished by checking the errors generated in TRIM against an upper limit and, whenever all errors are less than this limit, reducing both the partial derivative increment and the maximum amount which one of the TRIM variables can change in one iteration. This second entry to this subroutine, RATI, limits and applies the corrections to the TRIM variables.
- (15) <u>DATI</u>. This block data contains  $C_L$ ,  $C_D$ , and  $C_M$  tables for the NACA 0012 airfoil.
- (16) <u>DAT3</u>. This block data subroutine contains the Carta tables used in subroutine CMCALC.
- (17) DERIV. This subroutine evaluates the derivatives during a maneuver.
- (18) <u>DIFFER</u>. This is a subroutine of numerical differential technique. Outputs from it are quantities of velocities and accelerations required for the calculation of unsteady aerodynamic effects.
- (19) ERRCHK. This subroutine checks possible input errors in the program logic group which is the key input group.
- (20) EXTORS. It recalculates og location, inertias, and gross weight when any external stora is dropped. It also updates aerodynamic brake locations if brake deployment during a maneuver is exercised.
- (21) <u>FOCUS</u>. This subroutine is in the rotor analysis. It transforms fuselage quantities into rotor reference and calculates accelerations during a maneuver.

- (22) <u>FUSFNM</u>. As the name implies, this subroutine computes fuselage aerodynamic forces and moments. It also calculates rotor nacelle drag contributions.
- (23) <u>FUSINT</u>. This subroutine converts input array to mnemonics for the fuselage group. It also calculates cg location and inertias if external stores are included.
- (24) <u>GUST</u>. This subroutine is entered only during a maneuver in which a gust is being generated. It calculates the distance of each part of the rotorcraft from the start of the gust and then calculates from that distance the magnitude of the gust velocity at each point on the ship.
- (25) <u>HAPPES</u>. This is the core of Hamming's Predictor-Corrector Method. It does predictions, checks errors, and makes corrections. It also handles the integration of differential equations and calls other subroutines necessary to a maneuver.
- (26) <u>HARM. HARM1</u>. The harmonic analysis for blade loads at the trim point is done by this subroutine.
- (27) <u>HPCG</u>. This subroutine calculates the first three time points. It is used along with trim points to start Hamming's Predictor-Corrector cycles in subroutine HAMMS.
- (28) HRESP. The elastic modes during the quasi-static trim procedure are handled by this subroutine.
- (29) <u>INBLD</u>, <u>INBLD</u>, <u>INBLD</u>, <u>INBLD</u>. This subroutine converts input array to mnemonics for blade related data.
- (30) INIT. This subroutine initializes the arrays for the printout during a maneuver and also writes those arrays on disk for later plotting.
- (31) INRO. The function of this subroutine is the initialization and calculation of problem constants from the rotor inputs only.
- (32) INRTR. This subroutine initializes some of the rotor related data which are not done in subroutine INRO.
- (33) INSCAS. Initialization of the SCAS inputs is done here.
- (34) <u>INSTAB</u>. This subroutine initializes for a stability analysis. It calculates the partial derivatives needed for later computing the frequency response.
- (35) INTFRO. It interpolates blade natural frequency as a function of rotor rotational speed and blade collective pitch.

- (36) INVERS. This subroutine calculates the inverse of the mass matrix before the call to ALLMAT.
- (37) <u>IOMAT</u>. This subroutine prints the mass, damping, and stiffness matrices used in the stability analysis.
- (38) <u>ITRIM</u>. Included in this subroutine is the iteration loop of the trim section of the program. The function here is to iterate to a trimmed flight condition.
- (39) ITROT. This subroutine initializes variables for subroutine AZMUTH and, when specified by the input parameters, controls the iteration loops to balance the rotor flapping moments. It also calculates the trim conditions for an elastic blade.
- (40) JACOBI. As the name of this subroutine implies, its function is to calculate the Jacobian for use in the Newton-Raphson iteration method in TRIM or to calculate the displacement derivatives for use in the stability analysis.
- (41) JFBGIN, ATMINT. This subroutine converts input arrays to mnemonics for jets, flight constants, bobweight, and weapons groups.
- (42) JSTRED. This subroutine reads most of the input data groups.
- (43) <u>LGCINT</u>. The program logic group input array is converted to mnemonics in this subroutine.
- (44) LIZE. Initialization of some numerical constants is done in this subroutine.
- (45) LOADT. This subroutine computes and prints out the loads on an elastic blade at the trim point.
- (46) MAIN. This subprogram reads the control cards which direct the flow of the whole problem. The path is selected and calls are initiated to begin working the problem. Upon return, possible errors are checked for and appropriate action is taken. If an error is detected, an error message may be printed out. Then the program either terminates execution or starts the next problem, depending on the severity of the error.
- (47) MANU. This subroutine controls the time-variant maneuver segment if Runge-Kutta Method is used. It handles the integration of the differential equations and the calling of the other subroutines necessary to a maneuver.
- (48) MANUV. It controls the time-variant maneuver segment if Hamming's Method is used.

- (49) MATRIX. The function of this subroutine is to calculate the transformation matrix from a set of input Euler angles.
- (50) MBAL. This subroutine calculates rotor flapping increments during each iteration.
- (51) MDRDRS. Damping and stiffness matrices for a stability analysis are calculated here.
- (52) MNEM. The function of this subroutine is to calculate problem constants from input data and to initialize for a problem.
- (53) MODAL. The variables which are functions of mode shape, frequency, and mass and inertia distributions only are computed in this subroutine.
- (54) MODES. This subroutine calculates the mass matrix for a stability analysis.
- (55) MOMB, SUBA, SUBB. This subroutine simulates a servomechanism controlling the swashplate while the main rotor is being folded horizontally.
- (56) MTLT, TTLT, FLAT, YAWP. This subroutine handles mast tilt during a maneuver, the flat tracking mechanism, and the yaw control servomechanism.
- (57) NOPS. NOPS1. The inputs to this subroutine are the number of azimuth stations for use in the rotor analysis. The outputs are quantities which are functions of the number of azimuth stations.
- (58) NPUTOT. This subroutine prints most of the input data.
- (59) <u>NSTED. NSTEDI.</u> This subroutine calculates some variables for unsteady aerodynamic effects.
- (60) PDZ1, PDZ. The inputs of this subroutine are a trim partial derivative matrix (i.e., the Jacobian) and the type of helicopter or rotorcraft being flown. This subroutine then changes the partial derivative matrix to conditions which are known to hold. Essentially, this is an attempt to filter numerical "noise" in the matrix.
- (61) <u>PUNCH</u>. It punches nonzero elements of mass, damping, and stiffness matrices used in the stability analysis. The form of the punched output cards is explained in Volume II.
- (62) PYLINT. It converts input arrays to mnemonics for the dynamic pylon group.
- (63) <u>QUAN</u>. This subroutine sets mnemonics from the integration array at the end of each time point.
- (64) RADBGN. It calculates some variables used by subroutine RADIAL.

- (65) <u>RADIAL</u>. This subroutine in the rotor analysis does the calculations and integration along the blade radius.
- (66) RADOUT. It prints output from subroutine RADIAL and AZMUTH.
- (67) RBDY. It initializes radial stations for a rigid blade.
- (68) READIN. This subroutine contains the logic for reading and printing the input data.
- (69) REDATB. It handles the read-in of airfoil data tables.
- (70) <u>REDBMS</u>. It handles the read-in of aeroelastic blade data as well as blade mode shape data.
- (71) <u>REDCL</u>. It reads the coefficients of lift, drag, and pitching moment of each airfoil data table.
- (72) REDID. It handles the read-in of group ID cards.
- (73) <u>REDRWK, RWKOUT.</u> It performs the read-in and printout of rotor-induced velocity distribution (RIVD) tables.
- (74) <u>REDSWK, SWKOUT</u>. It performs the read-in and printout of rotor wake at aerodynamic surface (RWAS) tables.
- (75) <u>RESTRT. REST4, REST0, REST1, REST2</u>. Restart tapes are written or copied by this subroutine. For the first case of each run, it also initializes all variables in commons to zero.
- (76) <u>RGUST</u>. Subroutine <u>GUST</u> carries wind gust calculations to the rotor hubs. This subroutine then carries the calculations to the blade elements.
- (77) RNCKTA. This is a special Runge-Kutta routine being used as a starter in Hamming's Predictor-Corrector Method.
- (78) <u>ROTAN</u>. This subroutine may be considered to be the outer section of the rotor analysis.
- (79) RPTPG. .It controls the optional trim page output.

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- (80) RTINIT. This is the control routine which handles the initialization of the rotor.
- (81) RTWAKE. This routine calculates blade local induced velocity when rotor wake table option is used.
- (82) <u>SAVEIC, RESTOR, RESTR1</u>. This routine saves initial conditions during calculation of the first three time points of Hamming's Method.

- (83) SAVOLD. It saves the values of some variables at the previous time point for Hamming's Method.
- (84) SCASIT. The highest derivatives in the differential equations for the SCAS are calculated here.
- (85) SIVAR. This subroutine handles the initialization of the maneuver inputs for subroutine VARI which are not a function of the trim point.
- (86) <u>SOLVE</u>. This subroutine solves systems of linear equations by Gaussian elimination.
- (87) STAB. This subroutine computes the rate derivatives used in the stability analysis.
- (88) START. The function of this subroutine is to change units of the input arrays and set them equal to mnemonics and to control the initialization segment.
- (89) STBWAK. This subroutine calculates the effect of rotor wakes on each stabilizing surfaces when a surface uses RWAS tables.
- (90) STBZFM. It calculates aerodynamic forces and moments at all stabilizing surfaces.
- (91) STBZIN. The function of this routine is the initialization and calculation of problem constants for wing and stabilizing surfaces.
- (92) SUPERP. This subroutine simulates an autopilot.
- (93) SWAS. This subroutine performs the function of linking the controls to the swashplates with the appropriate linkage factors and phase factors.
- (94) SWSRAT. It calculates some intermediate velocities and accelerations.
- (95) TABFIX. This subroutine calculates arrays to be used in the method of calculated entry in subroutine TABINT.
- (96) <u>TABINT</u>. As the name implies, this subroutine does a table interpolation for  $C_n$ ,  $C_n$ , and  $C_M$  tables.
- (97) <u>TABOUT</u>. When tables are used for  $C_L$ ,  $C_D$ , and  $C_M$ , this subroutine prints the tables after the input data.
- (98) <u>TILT, HASF, TFFA</u>. This subroutine controls cg shift calculations for several different manners of shifting cg. The primary function is in a mast tilt maneuver. It provides not only for cg shift but also for changes in control phasing as a function of the mast tilt angle. Secondary entries handle cg shift with folding of a rotor either when it is

being folded aft after being tilted forward and stopped or being folded horizontally after a stop.

- (99) <u>TIMLP</u>. This subroutine contains the initialization necessary at the start of each time step.
- (100) <u>TIVAR</u>. This subroutine handles the initialization of the maneuver inputs for subroutine VARI which are a function of the trim point.
- (101) TRIM. As the name implies, this subroutine is the primary one of the section of the program for finding the trimmed flight condition.
- (102) TRMANU. This routine sets up arrays for the outputs of trim as well as maneuver pages.
- (103) TRM1. The primary function of this routine is to initialize variables used by time-variant trim.
- (104) TURN. This subroutine handles a banked turn. Secondarily, it handles pushovers or pull-ups. It does so by checking input data, picking up proper inputs, and doing the appropriate initialization to find a trimmed flight condition.
- (105) TVT. This subroutine controls the time-variant trim procedure.
- (106) <u>UNSTED</u>. This is the major section of the UNSAN unsteady aero-dynamic model mentioned in Volumes I and II.
- (107) <u>VARI</u>. This subroutine produces the effects of input disturbances during a time-variant maneuver. The inputs to this subroutine are the user-supplied forcing functions. The results produced by these functions are the output of this subroutine.
- (108) VCNTRL, VCNT1, VCNT2, ---, VCNT10. This is called by subroutine VARI. It calculates the effects of time-variant maneuver disturbance.
- (109) VIND. This subroutine calculates the induced velocity of a rotor.
- (110) <u>WAG</u>. The time-dependent lift change by the Wagner and Kussner Method is computed in this subroutine.
- (111) <u>WING</u>. As the name implies, this routine computes aerodynamic forces and moments on wings.
- (112) WKTABN. If the number of blade radial stations input to rotor wake table is not twenty, this subroutine interpolates the table into twenty blade radial stations. This is done outside the iteration loops so that a three-way interpolation can be reduced to two-way.

- (113) WRFM. This is an output subroutine which writes out the rotor force and moment summary in shaft reference and the fuselage reference force and moment summary.
- (114) WRINST, IOWRFM, IOWRFL. This subroutine prints output during the computation of partial derivatives for a stability analysis.
- (115) WRMANU. This subroutine contains the write statements which produce part of the maneuver printout.
- (116) WRMDAL, WRMDL1, WRMDL2. This subroutine prints the results of subroutine MODAL.
- (117) WRMS. As a result of stability analysis, it prints out mode shapes associated with the rotorcraft characteristic roots.
- (118) WROT1, WROT. This is another output subroutine which produces the heading for the printout of the input data and the trim page.
- (119) WRSTAB. This subroutine prints the rate derivatives used in the stability analysis.
- (120) WRTMNV. This subroutine defines the output arrays for trim as well as maneuver pages.
- (121) WRTRIM. As the name implies, this routine writes trim page.
- (122) WRVP. This is still another output subroutine which produces the printouts of the partial derivative matrices calculated and the independent variables used in the calculation of those derivatives.
- (123) WSHDUF. It calculates fuselage effects on downwash and sidewash angles at wings and other stabilizing surfaces.
- (124) XCONIN. Initialization of all control linkages is handled by this subroutine.
- (125) XSTINT. This subroutine converts input array to mmemonics for external stores/aerodynamic brakes.
- (126) XSTORE. It calculates aerodynamic forces and moments at each external store/aerodynamic brake.
- (127) YRINIT. This subroutine conditions the aerodynamic inputs for the rotors.
- (128) YSINIT. This subroutine conditions the aerodynamic inputs for the wing and stabilizing surfaces.
- (129) ZERO. This is part of the initialization segment. Every variable in this routine is set to zero.

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(130) ZLLCAL. This subroutine computes zero lift line increments at wings and other stabilizing surfaces.

# 3.3 ASSEMBLY LANGUAGE SUBPROGRAMS IN AGAJ73

AGAJ73 uses four assembly language subprograms which were written at Bell Helicopter Company: ABDUMP, DATE, DOTX, and PDSRED. The first three are available from the SHARE (an IBM user's organization) FORTRAN library. Although PDSRED itself is not available from SHARE, their library does include two routines (OPEN and CLOSE) which are generalizations of PDSRED and can be adapted to replace the single routine.

- (1) ABDUMP. This diagnostic routine permits the program to terminate itself with a memory dump under OS/360. It does nothing except issue an ABEND macro with user code 0322. The routine is in System/360 Assembler Language. It was prepared at Bell Helicopter and is in the public domain.
- (2) <u>DATE</u>. This routine with argument (NDATE) returns the current system date, NDATE, in Gregorian form: mm/dd/yy. NDATE must be at least eight bytas long. The routine is coded in System/360 Assembler Language for OS/36C only. It uses the macros peculiar to the MVT option of OS/360. It was prepared at Bell Helicopter and is in the public domain. It contains the following entry points:

ENTRY SETIME (TINT). This entry establishes an operating time interval against which to check program operation. This interval (TINT) is in minutes in floating point form. The routine does not cause execution to terminate at the end of the designated interval. This entry initializes TIMEX.

ENTRY TIMEX (TU, DT, TL). This entry checks the central processor time since the last call to SETIME or TIMEX. It returns three argument values in floating point minutes:

TU - Time since initial call to SETIME.

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- DT Time since last call to TIMEX or SETIME.
- TL Time remaining in the SETIME interval.
- (3) <u>DOTX</u>. This is a function subprogram with the following argument list: A, IA, B, IB, NP. It is a fast method of computing vector inner product using double precision accumulation of sum of products of single precision operands. The routine is physically small, about twice as fast as equivalent FORTRAN code, and quite flexible. It provides the improved accuracy of double precision vector products without the increase of storage and time required to do full double precision analysis. DOTX returns a double precision result if the calling program types it as such. The arguments are described as follows:

- A and B are the first elements of the two single precision vectors to be multiplied.
- IA is the number of words between successive elements of A. It is a full word integer greater than zero.
- is the number of words between successive elements ofB. It is a full word integer of any value.
- NP is the number of element products to be formed, i.e., the number of elements used. It is a full word integer greater than zero.

The only validity check made on the arguments is that IA and NP be greater than zero. If either of them is out of range, the routine returns a zero result. There are no other error exits.

The computations of the subprogram can be represented by the following summation:

$$X = \sum_{i=0}^{NP} Y(NA + i*iA)*Z(NB + i*iB)$$
 (3-1)

where

X is the output of the subprogram,

Y(NA) is the NA<sup>th</sup> element in the Y array (A in the calling sequence),

Z(NB) is the NB<sup>th</sup> element in the Z array (B in the calling sequence), and

NP, IA, and IB are as defined in the argument list.

For example, assume that the two arrays FORCE and DISP are dimensioned to 5 and 8 respectively. Using DOTX, the following statement could be programmed:

$$WORK = DOTX(FORCE(1), 2, DISP(2), 3, 3)$$
 (3-2)

The equivalent FORTRAN statement would then be as follows:

WORK = 
$$FORCE(1) + DISP(2) + FORCE(3) + DISP(5) + FORCE(5) + DISP(8)$$

The routine is written in System /360 Assembler Language for any level of assembler. It was prepared at Bell Helicopter Company and is in the public domain.

(4) <u>PDSRED</u>. This routine is used to find a member of a partitional dataset and to make it accessible to a FORTRAN routine through normal sequential READ statements. The parameter list is as follows:

member an 8-byte character string which identified the desired member

den the symbol & is a standard punch. The symbol on stands for a statement number. If the member is not found, the program goes to this statement in the calling routine rather than executing the statement following the call to PDSRED. This type of argument is an IBM FORTRAN language extension.

After returning from PDSRED, normal FORTRAN READ on unit 1 may be issued and an end of dataset is sensed as with any other dataset. Before another call to PDSRED can be issued for the same PDS, Remind 1 must be issued.

This routine is coded in IBM System/360 Assembler Language using OS/360 macro facilities. This function is not duplicated in any other operating system. The routine is compatible with all OS/360 releases through and including the 21.6 release. It was prepared at Bell Helicopter and is in the public domain.

# 3.4 FORTRAN SUBROUTINES IN GDAJ07

The FORTRAN subroutines contained in GDAJ07 are listed in alphabetical order including the main program, which is called MAIN. A few remarks are made for each subroutine which indicate its general use or purpose in the program.

- (1) <u>CALC81</u>. This subroutine is the interface between subroutine SCALIT and the CALCOMP plot routines.
- (2) <u>CONPLT</u>. This subroutine controls the plotting of the time histories.
- (3) <u>CURVET</u>. This subroutine analyzes the time history of selected variables during a maneuver. This analysis is accomplished by a least-squares curve fit followed by comparison of both the amplitude and phase angle of different variables. Then one variable is expressed as a linear function of two others.
- (4) <u>C81L</u>. The function of this subroutine is the transfer to a disk of maneuver time history data which has been stored on a tape or the transfer to a tape of maneuver time history data which has been stored on a disk.
- (5) <u>DATP</u>. This first block data subroutine contains part of the headings for plotted time histories.
- (6) <u>DAT1</u>. This second block data subroutine contains part of the headings for plotted time histories.

- (7) <u>FSFT</u>. This subroutine controls the harmonic analysis of a time history.
- (8) HARM. This is the harmonic analysis subroutine used by subroutine FSFT.
- (9) <u>HEDING</u>. This subroutine generates the labels for the time histories from the data stores in DATP or DAT1.
- (10) MAIN. This is the control program for GADJ07.
- (11) PLOTER. This subroutine does the CALCOMP plotting of the results of the harmonic analysis.
- (12) PPLOT. This is the printer plot routine which produces plots of time histories.
- (13) SCALIT. This subroutine sets up the arrays for the time history plots.
- (14) <u>SCLFIX</u>. This subroutine calculates scale factors for the time history plots.
- (15) WROT1, WROT. This subroutine prints the headings on the printer plots.

# 3.5 LABELED COMMONS IN AGAJ73

There are 21 labeled COMMONS, but no blank COMMON, in AGAJ73. Each of the COMMONS is listed below. Any special ordering of variables and the reason for doing so are given along with some general comments.

- (1) ANDOIT. The first 9 variables, HFRC through YSHRN, in this COMMON are double precision. It is used only in the general purpose subroutines controlled by subroutine ANAL.
- (2) <u>FLEX</u>. It contains most of the variables used in elastic blade model analysis.
- (3) FORCHC. This COMMON contains the Carta tables used by subroutine CMCALC. Since it is defined in block data format in subroutine DAT3, it is not marked as a COMMON block in Table 4-1.
- (4) <u>FORWK</u>. This COMMON contains most of the variables used in computing the rotor-induced velocity distribution from the table stored in FORWK1.
- (5) <u>FORWK1</u>. This is the set of rotor-induced velocity distribution (RIVD) tables used by subroutine RTWAKE.

- (6) <u>FORY</u>. There is no special ordering of variables in this COMMON. It consists of the array "Y" operated upon by Runge-Kutta and is used in the initialization, trim, and maneuver segments.
- (7) FORYYY. It contains most of the variables specifically used by Hamming's Predictor-Corrector Method.
- (8) <u>FORYYI</u>. This COMMON has only one array. It retains all the values of previous time points and their derivatives used by Hamming's method.
- (9) <u>FOSWK</u>. This COMMON contains most of the variables used in computing the effects of the rotor wake at the aerodynamic surfaces from the tables stored in FSWK1.
- (10) <u>FOSWK1</u>. This is the set of rotor wake at the aerodynamic surfaces (RWAS) tables used by subroutine STBWAK.
- (11) INSTAR. This block contains most of the input. It is used in the initialization segment.
- (12) MANAL. The first 59 variables, XF through NQTR, in this COMMON are ordered to allow I/O and other manipulations to be done on an equivalent array. The next 11 variables, COLSTK through BlTR, are ordered for equivalencing to an array. Not more than 10 of these variables are used, and the array KVAR is used as a pointer vector to choose which ones and their order. The next 22 variables, AlM through PYLMD, are ordered for equivalencing to the array VAR in subroutine STAB for the calculation of derivatives. The variables TAXL and TAXR are equivalenced to an array in subroutine CNTM.
- (13) <u>STAMAN</u>. Variables in this block are mostly used in initialization and maneuver segments. Arrays SHPGRP through SFTGRP are ordered to allow I/O and other manipulations to be done on an equivalent array.
- (14) STARAD. Most of the variables here are used in the initialisation and general-purpose segments.
- (15) STARAN. This COMMON is used in the initialization and general-purpose segments.
- (16) STBD. This COMMON is used only in the stability analysis.
- (17) STRIAB. COMMON STRIAB is used in the initialization, trim, and stability analysis segments.
- (18) STRIMA. The first 16 variables, TZM through TCLOCK, are ordered for equivalencing in subroutine MOMB. The next six arrays, SCASRF through SCASYC, are ordered for equivalencing in subroutine INSCAS. This COMMON is not used in the rotor analysis.

- (19) TAB. This COMMON contains the C table for the NACA 0012 airfoil. Since it is defined in block data format in Subroutine DAT1, it is not marked as a COMMON block in Table 4-1.
- (20) <u>TABl</u>. This COMMON contains  $C_L$  and  $C_D$  tables for the NACA 0012 airfoil. Like TAB, it is block data in DATl. Consequently, it is not identified as a COMMON block in Table 4-1.
- (21) TOPLOT. This COMMON is used in all segments.

# 3.6 LABELED COMMONS IN GDAJ07

There are 6 labeled COMMONS, but no blank COMMONS, in GDAJ07. Each of the COMMONS is listed below, together with pertinent comments.

- (1) <u>INPLOT</u>. This COMMON is used by subroutine SCALIT and the other subroutines in the segment for plotting time histories.
- (2) MAXMIN. It contains the maximum and minimum values of the specified variable. It is primarily used to determine the scale of the plot.
- (3) <u>PLOTD</u>. This COMMON contains the data in the block data subroutine DATP which is used by subroutine HEDING to furnish alphameric headings for time histories.
- (4) PLOTD1. It has the data in the second block data subroutine DAT1 which is used by subroutine HEDING to supply headings for time histories.
- (5) TIMPTS. This COMMON is chiefly used as local storage in subroutines CURVET and C81L.
- (6) TOPLOT. This COMMON contains control variables and is not the same as COMMON TOPLOT in AGAJ73.

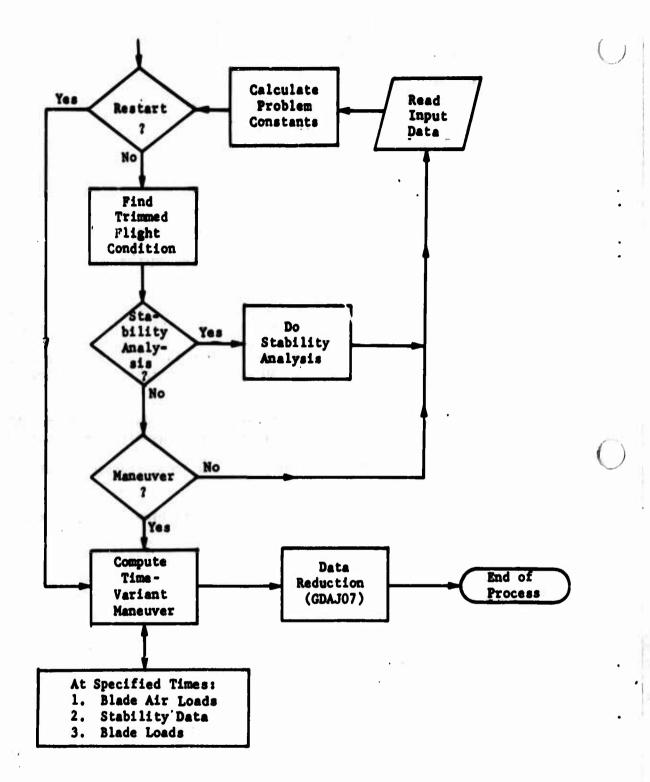


Figure 3-1. Flow Chart of System Structure.

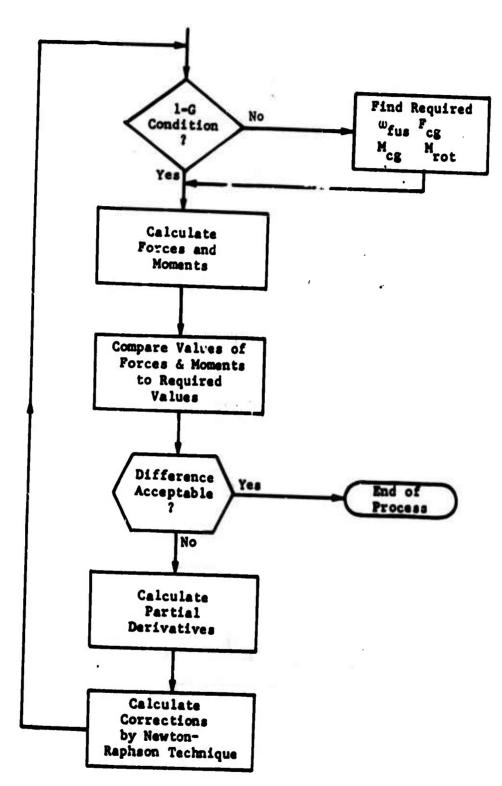


Figure 3-2. Flow Chart of Trim Process.

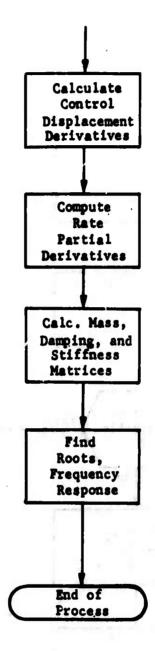


Figure 3-3. Flow Chart of Stability Analysis.

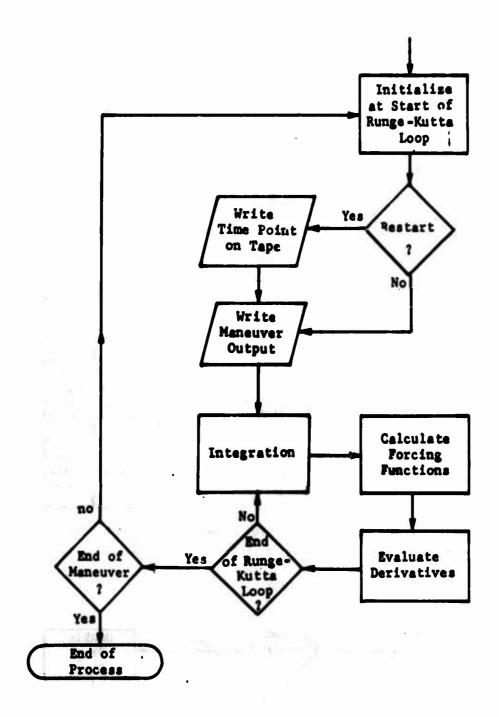


Figure 3-4. Flow Chart of Maneuver With Runge-Kutta Method.

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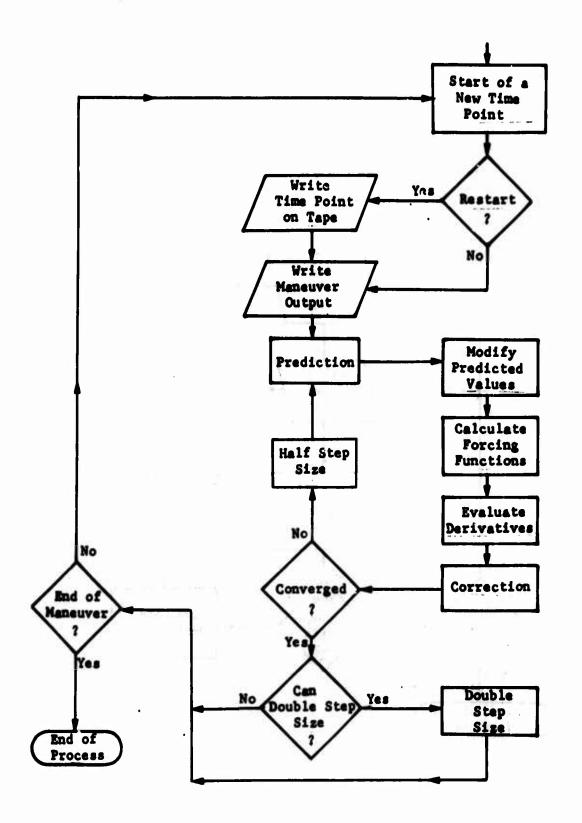


Figure 3-5. Flow Chart of Maneuver With Hamming's Method.

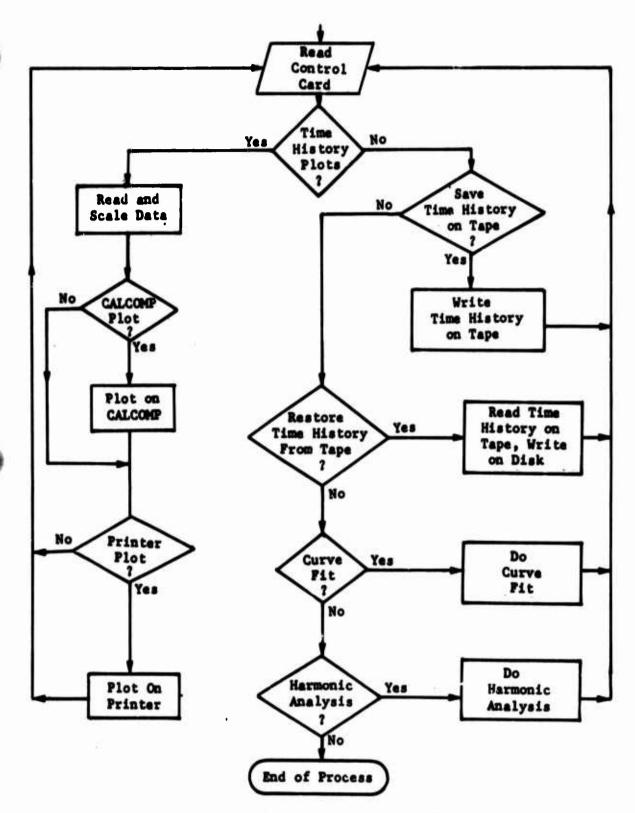


Figure 3-6. Flow Chart of GDAJ07.

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## 4. DETAILED PROGRAMMING AIDS

#### 4.1 CONTROL SECTION CROSS-REFERENCE

The Control Section Cross-Reference List for AGAJ73, Table 4-1, shows most of the control sections, including COMMONS, which are referenced by another control section. The exceptions are listed on the first page of the table. These exceptions are system routines, and their presence would not contribute to the cross-reference list. The remaining pages of Table 4-1 contain the control sections in alphabetical order in a column on the left side of the page. To the right of each control section name is the cross-reference information. "LENCT!" is the size of the subroutine or COMMON in hexadecimal bytes. "CALLED BY" gives the name of each control section referencing the control section whose name is in the column on the left. "IS USED BY" gives the name of each control section referencing a control section in the "CALLED BY" list or by another control section in the "IS USED BY" list. "CALLS" gives the name of each control section referenced by the control section whose name is in the column on the left. "USES" gives the name of each control section referenced by a control section in the "CALLS" list or by another control section in the "USES" list.

The information in the Control Section Cross-Reference List is sufficient to construct the sequence of subroutine calls from which an overlay structure can be made.

As noted in Section 3.2, several subroutines have multiple entry points. However, the Cross-Reference List (Table 4-1) includes only the primary names of subroutines; it does not include the names of any of these additional entry points. In the case where a call to a subroutine is actually a call to an additional entry point, the primary name of the subroutine which contains the specified entry point is used in the Cross-Reference List. For example, Table 4-1 indicates that subroutine AZMUTH calls RADOUT when AZMUTH actually calls AZMOUT (a second entry point to RADOUT). In Section 3.2 the names of any additional entry points follow the name of the subroutine; however, in reading the program listing, it is generally more useful to know the name of the subroutine which contains a particular entry point than to know the additional entry points of a subroutine. Hence, a cross-reference of the additional entry points to the subroutine which contains them is given in Table 4-2.

Table 4-3 contains the Control Section Cross-Reference List for GDAJ07. It is read and used in exactly the same manner as Table 4-1.

#### 4.2 PROGRAM LISTINGS

Listings of the three programs supplied under this contract are on file at Eustis Directorate, USAAMRDL:

- (1) AGAJ73
- (2) GDAJ07
- (3) AS812A

Note that Program AS812A is written in PL/1 rather than FORTRAN.

Program AS812A is independent of the AGAJ73/GDAJ07 combination. The purpose of the program is to perform least-squared-errors curve fits of wind tunnel data which provide inputs for the Nominal Angle Equations of AGAJ73. The card format for the wind tunnel data input to AS812A was chosen to be compatible with the format of data supplied by the LTV Low Speed Wind Tunnel, Dallas, Texas. The data is read in by the statement on CARD 500 for the sequenced source deck. The format may be changed by the programmer to read most any available data.

The printed output of AS812A lists the coefficients of the equation to which the data has been fitted. The output also lists the inputs to AGAJ73 in the proper sequence and format with the AGAJ73 card number. Punched card output, for direct input to AGAJ73, is an option in AS812A. The job control cards for AS812A are shown in Figure 4-1.

# 4.3 PROGRAMMING OPTION IN NUMERICAL INTEGRATION

As mentioned in Section 2 of Volume I, there are two methods of numerical integration techniques in AGAJ73. The first one is the Runge-Kutta method which is programmed as the default option. The second is Hamming's method which can be activated only by setting the logical variable HMG to TRUE. This variable is located in subroutine LIZE and is set to FALSE in the program delivered under this contract.

# 4.4 REDUCTION OF PROGRAM MEMORY REQUIREMENTS

There are several very large arrays in AGAJ73. The size of some can be easily reduced without any significant decrease in the capabilities of the program. The arrays which can be easily reduced in size are shown in Table 4-4. The table also includes the name of the COMMON to which each belongs, the full and reduced dimensions of each, and the approximate savings in memory requirements for reducing each array. Up to 330K of storage can be eliminated with minimal effort.

Referring to Table 4-3; arrays WKRTR, WKSTB, and AUX are independent of each other. Their sizes can be cut individually or concurrently. However, arrays CURVED, CURVEL, and CURVEM are dependent; if one of them is reduced, all of them must be reduced at the same time. If, after the program is link-edited in load-module form, the programmer wishes to reduce the size of any array noted in Table 4-4, it is necessary to include a REPLACE card, before the INCLUDE OLDLIB or SYSLMOD card in the linkage step. This REPLACE card must then contain the names of all arrays which have just been reduced in size.

In Table 4-4 there are two columns under the headings of array name and dimension. The "Outride RESTRT" column gives the name(s) and dimension(s) to be used in the associated COMMON block when it is included in any subroutine other than RESTRT. In subroutine RESTRT the information listed under "In RESTRT" must be used. The different names and single dimensions are in RESTRT simply to facilitate the task of setting the COMMON blocks to zero. Each array included in Table 4-4 is discussed in more detail below.

WKRTR contains the sets of coefficients for the Rotor-Induced Velocity Distribution (RIVD) tables discussed in Sections 2.3.3.1.3 and 2.3.3.2 of Volume II. If the size of this array is reduced as shown in Table 4-4, the input to IPL(5) must be equal to zero and in subroutine RESTRT FRK1(66000) and DO 34 I=1,66000 must be changed to FRK1(1) and DO 34 I=1,1, respectively.

WKSTB contains the sets of coefficients for the Rotor Wake at Aerodynamic Surface tables discussed in Section 2.3.4 of Volume II. If the size of this array is reduced as shown in Table 4-4, the input to IPL(37) must be equal to zero and the inputs to XWG(29) through XWG(32), XSTBi(29), and XSTBi(32) must all be less than or equal to 100. XSTBi stands for the i the stabilizing surface. In subroutine RESTRT, FSK1(9000) and DC 36 I=1,9000 must be changed to FSK1(1) and DO 36 I=1,1, respectively.

CURVED is the array where drag coefficients of the airfoil data tables are located. If the size of this array is reduced as shown in Table 4-4, the input to IPL(2) must be less than or equal to 2, TABL(6350) in subroutine RESTRT must be changed to TABL(3026), and TITLE(8,4) and CURVED(1100,4) in block data DAT1 must be changed to TITLE(8,1) and CURVED(1100,1), respectively.

CURVEL and CURVEM are the arrays for the airfoil lift and pitching moment coefficients respectively. If their sizes are reduced as shown in Table 4-4, the input to IPL(2) must be less than or equal to 2, TABL1(5375) in subroutine RESTRT must be changed to TABL1(2150), and CURVEL(1100,4) and CURVEM(575,4) in block data DAT1 must be changed to CURVEL(1100,1) and CURVEM(575,1), respectively.

AUX is an auxiliary array used in the Hamming's method. If this array is reduced as shown in Table 4-4, the default option of numerical integration in maneuver (the Runge-Kutta method) must be used. Also, in subroutine RESTRT, FYY1(3440) and DO 104 I=1,3440 must be changed to FYY1(1) and DO 104 I=1,1; respectively.

There are other large arrays which can be reduced if storage requirements are still excessive after taking the steps discussed above. However, the necessary steps for reducing them are more complicated and in some cases mean deletion of a major program option, and in most cases the savings is not worth the effort. The programmer should consult with the contractor before attempting any storage reductions beyond those outlined in this report.

## 4.5 SWITCH FOR DIAGNOSTIC DATA FROM STAB

In Section 3.2 of Volume II, IPL(34) is defined as a switch for obtaining diagnostic data during the stability analysis (STAB). Since the data generated by this switch are not of general interest to the user, but can be useful to the programmer, the function of IPL(34) is discussed in this Programmer's Manual rather than in Volume II. The function of the switch is described below.

In STAB there are up to 22 independent variables which may be incremented in the process of computing the stability (partial) derivatives. The number of variables actually incremented depends on the number of degrees of freedom which the user has activated. (See IPL(30) and (32) in Section 3.2 of Volume II). In each STAB case, IPL(34) can be used to print out the following data resulting from one of the variables being incremented:

- (1) blade element aerodynamic data ( $\alpha$ ,  $C_L$ ,  $C_D$ ,  $C_M$ , etc.) at each blade station and each azimuth location for each rotor (i.e., IPRINT in subroutine RADIAL does not equal zero, which calls RADOUT)
- (2) rotor and pylon moment data (i.e., COND1 in subroutine MBAL is greater than 1.5, which causes printout)

To generate this output for a particular increment, IPL(34) is set to the value shown in Table 4-5. Further information about the variables in this table can be found in Section 4.11.2.1 of Volume II. Note that locking out a degree of freedom does not change the correspondence shown in Table 4-5 between IPL(34) and the variables. Also, it is only possible to obtain this extra printout for one variable in each STAB case. To obtain the printout for more than one variable, the case must be rerun for each variable of interest with IPL(34) set to the appropriate value in each repeat run.

CONTROL SECTION CROSS-REPERENCE FOR AGAJ73	CONTROL SECTION REMOVED
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	STRINA FOCUS INTERES STANAM	•	ESTORE FOCUS TTROT STALLS	TABINT TABINT MARMS MOTAN	MINI.	TOPLOT RESEA
	STR 1AB FLEX INSTAR RTVAKE TOPLOT		MANU MING FLEX STAND STAND	MESS SYSUAT POCUS MESTA	BUSTAB	INSTAB STABAN PORMEL TABINT
	STARIA BOTS BOTS FOTAN		MAIN TYPELOT ROTX STARAN UNSTED	FUSPIN STRZFII DERIV RADIAL	Š	STACAD STACAD FORM TAB
	STAND DIPFER PUSTM REUST TABUST		JACORI STRINA DIFFER HRESP RTMAKE	FOCUS RPTP6 CONTRI	HANG TRIN STARAGE	HADINE TYT RABOUT FORCIC STR JUA
Continued.	Thin Astrik Cocald Fosser		178 IN 518 2 FM 518 2 FM 50 5 MK 1 66 US 7 10 PL 0 7	CHCALC ROTAN CONSTB NAMELY	FICUS STAB	POCUS TRIP RADIAL PLEX STARAR
	MAIN MAIN MAIN MAIN MAIN MAIN MAIN	ž	INSTAB STARAN CLCD POSMK RADCUT TABI	CLCO RAME	DERIV ROTAN RABAL	STAP STAP NAMA BOTX STARAD
TABLE 4-1.	ITRIH FORVYI COCL FORVYI STRIMA STRIMA	oets	MPCG ROTAN CDCL FORTYI RADIAL TABINT	IOL SECTIC CDCL RADOUT CDCL MAIN	CONTRA RINGK TA FORMK 1	CONTRR NOTAN INTERO DIFFER STANAN
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	4800mm	ALUMAT	<b>1</b>	<b>1100H</b>	AZHENT	42 42 44 44 44 44 44 44 44 44 44 44 44 4

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	UNSTED ANAL MAIN MANAL TAB	TILT HAMMS MANAL	WING ANAL RNGKTA MANAL TAB	ANAL MAIN FORCMC	HAMMS	INSTAB ALLMAT FLEX INSTAR RADBGN STRIAB	STRIA6 ANAL FOCUS INSTAR RADIAG STRIAG UNSTED	ITRIM
	AZMUTH MANU STARAD TAB1	HPCG	CONSTB STAB STARAN TAB1	AZMUTH	HPCG	MANAL ANAL FOCUS INTFRO RADIAL STRIMA WRFM	TOPLOT ANDOIT FORCMC INTERO RADOUT STRIMA VIND	TRIM
	CONSTB MANUV STARAN	MAIN	CONTRH TRIM STRIAB	CONSTB	MAIN TOPLOT	MODES ANDOIT FORCMC INVERS RADOUT SWAS WRINST	TRIM AZMINT FORWK ITRIM RGUST SWAS MAG	
	CONTRM RADIAL TABINT	MANU	DERIV	CONTRM	MANU	STAB AZMINT FORWK IOMAT RGUST SWSRAT WRMS	AZMUTH FORWKI ITROT ROTAN SWSRAT WING	
	DERIV	MANUV	HAMMS	DERIV	HANUV	STARAN AZMUTH FORWKI ITROT ROTAN TAB WRSTAB	CDCL FORY JAPPG APPG TAB	
	FOCUS	MNEM	ЭЭ	FOCUS	RNGKTA	STBD CDCL FORY JACOBI RTWAKE TABINT	CLCD FORVYY MANAL RTMAKE TASINT WROTE	
	HAMMS	MOMB	INSTAB	HAMMS		STRIMA CLCD FORYYY MANAL STAMAN TABI WSMDUF	CMCALC FORYYI MATRIX SOLVE TABI WRTMNV	
	TRIM	MTLT	ITRIM	17T		TOPLOT CMCALC FORYYI MATRIX STARAD TOPLOT XSTORE	DAMPER FOSWK MRAL STAMAN TIVAR WRTRIM	
	INSTAB	RNGKTA	JACOBI	INSTAB		DAMPER FOSMK MEAL STARAN TVT ZLLCAL	DATE FOSWKI NOPS STARAD TOPLOT MRVP	
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## ## ## ## ## ## ## ## ## ## ## ## ##	LENGTH 980 CALLED BY - START IS USED BY - MAIN CALLS - 2MSCA USES - STAMA	0 START - RAIN - INSCAS - STARAM	INSTAR	RANAL	MATRIX	STANAN	STARAD	STARAN	STRIA	STRINA	101401		
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<b>1006</b>	CALLED BY - 15 CALLS - CALLS - USES - CALLS - USES - CALLS - USES - CALLS - CA	CCMSTB MAJN IONAT DATE	HAHAL INSTAR	PORCAS	PURCH STANGE	STARM	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	SAIRS	STRICE	TOPLOT			
Ş	LENGTH 878 CALLED RY - 15 USED BY - CALLS - USES - CALLS - CALLS - USES - CALLS - CALL	78 - VAR1 - FORV - CC2 ABR	HAMES FORYTY INSTAR	HPCG PORYYI RAMAL	PA378 PABBAL STANAS	Ser.	SOLVE TOPLOT	STANS STANS STANS	STRIPLE	r. tr	. 101-01		
136	LENGTH 846 CALLED BY IS USED BY CALLS USES	- VARI - DERIV - HANGL - CG2ARR	HAMES PATRIX INSTAR	STANAS STANAS MAIA	MAJE STRIPA STARKE		POST	MENTA					
•	CALLED BY - 15 USED BY - CALLS - CALLS -	- MAEN - CONTRA - NAMAL	8 A A A A A A A A A A A A A A A A A A A		2	i	į	į		START	i i	, mar	

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		\$ \$					TOPLOT	FPC6 TVT STARAB	STANA	MAR	3
		14					STRIMA	HAIR TAIR STARAB	25 T 20 20 20 20 20 20 20 20 20 20 20 20 20	FORMET	S RIST
	ı	2008 2012	START				STANA	FOCUS STAR STARAKE	PATER TRIR RADOGE	FORM	FOCUS
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TABLE 4-1.	STABLES	CORSTS RAMEN STARAM	7847		8	STANS	FORVE	CONSTB PARKEY FORWALL FORWALL	CONTRA RINCKTA FLEX	X100 1441	COMSTB RAMAN STARAN
F	1	22.00 SANTE	RE 46111	Ę	STANLE	START	RAMU PA IN FORVYY	AZRUTA PARU FORME FORME	CONSTR	DIFFER TABINT	AZMUTH HAMU STARAD
	START INSTAR TOPLOT	AMAL PA.14 DIFFER	17	11	MA IN INSTAR	RTINIT	HAMES FORV	ANAL NATU FLEX DOTX	MAA. COCL	188 188	PROINT PAIN PAIN PAIN
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TABLE 4-	PORMET REDSIER	\$		START	REDSUK START	AE010	REDIO	HANDY HAND FORY STAINA STARAN	CONSTB NAMEY STARAN
START	RANAL FORUM REDWIN	STAT	START	AEAD TH	REDAIK	START	START	HAIN FORWEI STRIAB STANAN	AZMUTA PAMU STARAD
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LENGTH 130 CALLED BY - JSTRED 15 USED BY - MAIN	LENGTH DIS CALLED BY - MAIN IS USED BY - MAIN CALLS - INSTAR USES - DATE REPCL	LENGTH 2FB CALLED BY - JSTRFD IS USED BY - MAIN TALLS - INSTAR USES - INSTAR	LENGTH 1FO CALLED BY - JSTRED IS USED BY - MAIN	LENGTH 2CO CALLED BY - REDATE IS USED BY - JSTRED	LEMETH 640 CALLED BY - JSTRED IS USED BY - JSTRED CALLS - INSTAR	LENGTH 1180 CALLED BY - JSTRED 1S USED BY - MAIN CALLS - FORM USES - INSTAR	LENGTH CFB CALLED BY - JSTRED IS USED BY - MAIN CALLS - FOSW USES - INSTAR	LENGTH 1058 CALLED BY - INIT IS USED BY - HARMS CALLS - FLEX STARAD USES - INSTAR	LENGTH 300 CALLED BY - AACON IS USED BY - AACON JACON CALLS - ANDOIT
REGY	1E AD IN	REDAT8	4ED8#S	REDCL	af010	RESE	REDSUK	RESTRY	REUST

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AZ PATTA COCL FORMS FORMS 1707 HANA REUST ROTAN SUPERP SWAS VIND WAG	STALAN TOPLOT COCL COCALC MRESP INTRO	MARKAL STANAM	STAMAN STAMAD INSTAM MANAL	CONSTB CONTAN	SAVOLD STANAM STANAM STRINA	SAVEIC HAMD HAMUV STAMAN STRIMA	
CGZARM FORV MATRIX STSAKE SUSRAT WING	TVT TVT DIFFER STARAM	STARAD	STARAN	DERIV	STRIM	START	
CLCD CHCALC FORTYY FORTY] HAAL HOW SCASIT SOLVE TAB TABINT	INSTAB ITRIN DOTK FLEX MANAL MATRIY STRINA BUSRAT	STARAN STAIAB	STRIAD STRIPA MODAL PTRIFE	FOCUS MANUS ROTAR STAB		TIME	
11 POSSE MT(7 STANAN 11A81 E 11A81	1 140081 17 148 17 148	18 STRIMA	LA TOPLOT	. 25 E			
DIFFER FOSMI MOPS STARAD TILT	HAIN FORCAC NSTED TABINT	E COTI	STARAD	INSTAR			
DOTX FUSFUM NSTED STARM TOPLOT	FORME FASSES TABL		STABAN	121			
		T-Day	STARAD . STARAH STRIMA				

					2	TABLE 4	4-1. C	Continued	đ.					
STVAR	CALLED BY 15 USED BY CALLS	E .5	- 1046 P - 1047 P - 1057 AR	RESTRT 100CG NANAL	THIT	MAIN STARAN	HANN STR INA	RANCO TOPLOT	¥	START	TINLP			
SOLVE	CALLED BY 15 USED BY CALLS	Ş' , S	428 - 17RIM D BV - CONTAM CALLS - STRIMA	DERIV TOTOL	SUPERP	5	. MATA	200	A STATE OF THE STA	PREKTA	Ē	TAAV		
<b>8</b> 72	LENGTH PACELLED BY CALLS CALLS	\$ 1111	CCKSTB - MAIN - AMAL - AMAL - AMAL - AMAL FOAWII WASTRIX MATRIX MATRIX STED VIND	MANAL PORV PORV STBUAK MAG	STARAN AZRUTH FORVVV NSTED ST8ZFH WING	STED CDCL FORYYI RADECH STRIAB	STR 1 AB CLCD FOSM RADIAL STR INA	STR.1MA COCCALC FOSWRI RABOUT SWSRAT KSTORE	SMAS DIFFER FUSFUM RGUST	TOPLUT BOTX MEESP ROTAN TABINT	ME INST FLE N INSTAR RTWAKE TABI	WASTAB FRCUS INTERQ STANAN TOPLOT	ZLLCAL FORCHC 1780T STARAD TVT	FORME NAMAL STAMAN UNSTED
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STALLO	CALLED BY	§' '		A.	MON* CONT COCL MPUTOT STRZFH AZMUTH JACOBI START	MON* CONTROL SECTION COCL FOCUS MEUTOT NSTED F STAZEM TRAAMU UA AZEMUTN CONSTR CA AZEMUTN TONSTR CA START TIME T	HRESP PYLINT UNSTED CONTIN MANU TRIM	HABLD PADBON VIND DERIV HABOV TVT	TMCO RADIAL WING FOCUS MEN UNSTED	INRTR RACOUT WATE IN RADOCH WATE IN	INSTAB READIN HPCG RADIAL	ITROT RESTRT INIT READIN	JFBG IN YGUS T INBO RBGKTA	JSTRED BPTPC INSTAN
STARM	CALLED BY	2	THIS ALACOS HAESP LLOADT TREIL ALACOS INMO RADBGM UNESTED	1S A "COMMON" CONTROL SECTION ANAL AZMINT AZMUNH COC INBLA NORDES MMEN HOD FOLA MORDES MMEN HOD ROTAN RPTPG NTINIT SIV AVASTAB ANAL RADIAL RESTRT RMCKTA ROT VARI NING MATRIN	MON* CONT AZATAY AZATAY AZATA MONDOS NOSTE ANSTE ANSTE ANSTE ANSTE ANSTE ANSTE ANSTE ANSTE ANSTE ANSTE	AZMUN MEN MEN MEN MEN MEN MEN MEN MEN MEN ME	IDN COCL INSTAB MODAL SIVAR SIVAR CONSTB LACON	CLCO INTFRO HODES STAB STAB CONTIN LIZE RTINIT	CONSTR ITAIN NSTFO START DERIV NAIN STAB	DAMPER 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107 17107	Freus Jaces Radber Strill Strill Pusser Radev Strien	TUST NATIONAL STREET NATIONAL	FUSING RATIOUT TAILS PACE PECE PECE PECE PECE TRIB	CUST LIZE NESTAT THEMSO THEMSO THE MODES

				TA	TABLE 4-1.		Continued.						
	LENGTH 84 CALLED BY 15 USED BY CALLS USES	PAO - VARI - DERIV S - PANAL - STRINA	HAMES SOLVE TOPLOT	HPCE STARMS	PAIN STRING	RAME TOPLOT	Anapra	RACKTA				•	
SMAS	15 USED BY - CALLS - CALLS - CALLS - CALLS	AJACOS CONSTS START	PREH CONTAN TRIN STAPAN	STAB DERIV STRIMA	VART	93 <b>1</b>	INSTAB	#I #I	JACO <b>8</b> I	# 1	3	2	AFTA
SERAT	LENGTH 740 CALLED BY - 15 USED BY - CALLS -	740 - FOCUS - FOCUS V - AJACOB 12 - AMDOIT	ANAL ENCKTA FLEX	CCNSTS ROTAN FORME	CONTRH STAB FORUKI	DER IV TR IM MANAL	HAIMS TVT MATRIX	PPC6 STARAN	INSTAL	Ĕ	JACOBI	į	3
\$	LEMCTH 6330 CALLED BY IS USED BY	- ALCOATS - AJACOS INSTAS ROTAN	RESTRY ANAL ITRIM STAD	TABINT AZMUTH ITROT START	TABOUT CDCL JACOBI STBZFH	CLCD JSTRED TIME	CONSTB MATH TRIM	CONTRA PLANE	DERIV MANUV UNSTED	FOCUS MEEN	RADIAL	MECC MECC	INIT BECTA
TABETX	LENGTH 3DB CALLED BY - TAPOL IS USED BY - MAIN	3DE - TAPOUT	STANT										
TABINT	15 USED BY 15 USED BY CALLS CALLS	858 CDCL - AJACOBI WING 15 - ANDVIT	CC PACE PACE TAB	AZMUTH MANU TABI	COMS TO	CONTRA	DER IV	ROCUS	STAB	97 <b>9</b> 15	3057AB TR.I.R	H. F.	1700T UMS TEB
14804	CALLED BY 15 USED BY CALLS	740 BY - START D BY - PAIN CALLS - INSTAR	\$	TABFIX	1481						16		
74	LENGTH 5400 CALLED BY 15 USED BY	00 - AJACOB 105 7AB ROTAN	MESTRY ANAL ITRIN STAB	TABINT AZMUTN ITROT START	TABOUT COCL JACOB1 STB2FH	CLCD JSTRED TIMLP	COPESTS	See The See Th	DERIV KLINIV UNSTED	FOCUS MINEP	NAMES RADIAL	E 60 T	HIT
7317	LENGTH SOCIALIS TO CALLED BY TO CALLS TO CALLS	900 - meen - Deriv - Celara - Instan	TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE	MTLT 19CG STANN STANN	VALI STRIB STRIB	TOPECT TOPECT	RANDY	BECKTA	START	YARI			

	RESTRT		FOCUS LIZE READIN SUPERP WRMANU	INSTAB NPUTOT TILT	STRIAB	HRESP RADIAL STRIMA WING			
	MANAL		EXTORS LGCINT RADOUT STBZFM MING	INIT	STARAN	FUSFNM RADBGN STRIAB WAG			
	INSTAR	1146	ERCHK JSTRED RADIAL START VCNTRL	HPCG MDMB STBZFM	STAMAN	FOSHKI PDZ1 STBZFM VIND	TOPLOT		
	FOSHK1 TAB1	START	DERIV JFBGIN QUAN STAB	HAMMS MODES START	RPTPG	FOSKK NSTED STBWAK UNSTED	STRIMA	TOPLOT	
	TOPLOT FOSWK TAB	MNEM	CONTRM JACOBI PUNCH SOLVE UNSTED	FOCUS MNEM STAB	ROTAN	FORYI NOPS STARAN TVT	TRIM	STRIMA	
-	STRIMA FORYY1 STRIMA	MANUY	CONSTB ITROT MTLT SIVAR TVT	EXTORS MANUV RTINIT WRTRIM	1204	FORYYY MBAL STARAD TRMANU	TTHE	STRIAB	
	STAMAN FORYYY STRIAB	HAND	CNIM ITRIM MOMB SCASIT TRM1	DERIV MANU RPTPG WRMANU	MANAL	FORY MATRIX STAMAN TOPLON XSTORE	HANUV	STARAN	
	HANAL FORY STARAN	HAIN	CLCD INSTAB MNEM RTINIT TRMANU	CONTRH MAIN ROTAN WING	HPTRIM	FORWKI MANAL SOLVE TABI WSHDUF	MANU	NOPS	
	INIT FORWKI STARAD	TINI	ANAL AZMUTH CGZARM C HPCG INIT INRO I MANU MANL M MANU MESTRY ROTAN R TIMLP ITVAR TRIM T	CONSTB JSTRED RNGKTA VARI	FORYYI	FORWK JACOBI RTHAKE TABINT	MAIN	HANAL	
	MANU MANUV FORYYI FORMK STAMAN	HPCG TOPLOT	AZHUTH AZHUTH INIT MANUV RESTRI	AZMUTH JACOBI RESTRT	FORVY	FORCAC ITROT ROTAN TAB	HPCG FORYY1	FORYYI	
-	HPCG MAIN FORYYW FLEX SIVAR MRMANU	TRIM HAMMS STRIMA	ANAL HPCG MANU REDSWK TIMLP	ANAL ITROT READIN TRIM	FORY	FOCUS INTFRQ RGUST SWSRAT WROTI	MRTRIM HAMMS FORYYY	MAIN FORVY	START
	LENGTH 6B8 HAMMS CALLED BY - HACG IS USED BY - HPCG CALLS - FORY USES - DATE USES - DATE	LENGTH 320 CALLED BY - RESTRT IS USED BY - CONTRM CALLS - MANAL	CALLED BY - AJACCB CALLED BY - AJACCB HAPMS MANN REDNW REDNW TILT	IS USED BY - AJACOB ITRIM RADIAL TIMEP	LENGTH B18 CALLED BY — CONTRM IS USED BY — MAIN CALLS — DAMPER STRIMA		LENGTH EBO CALLED BY - INIT IS USED BY - CONTRM CALLS - FORY	LENGTH A20 CALLED BY - TRIM IS USED BY - CONTRM CALLS - FORY USES - MANAL	CALLED BY - MNEM IS USED BY - MAIN
	TIME	TIVAR	TOPLOT		TRIM		TRMANU	TRMI	TURN

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LENGTH CO CALLED BY 15 USED BY CALLS USES	COS - RADIAL - AJACOS JACOS - AMOOIT	T COCC	AZMUTH MANU MANAL STARAD	CONSTB PANUV STARAD STARAN	CONTRH RHCKTA STARAN TAB	DER IV ROTAN TOPL (T	FOCUS STAB TAB1	MANUS R I R I	95±74	INSTAB	<b>.</b>	i i
CALLED BY IS USED BY CALLS USES	EDE DERIV 15 - DERIV 15 - CNTR 5 - CGZARA STRIAD	EXTORS EXTORS P SUAS M FLEX	MAIN FLEX TILT FORY	#4#U FORY TOPLOT FORVYT TOPLOT	MANUV FORVYV VCNTAL FORVYI ZLLCAL	RMEKTA FORVYI ZLLCAL ZNSTAR	T SUS T	PARAL PATRIX	2 6	re Reg	ធធ	STANA
CALLED BY IS USED BY CALLS	DEO YAII - OFRIV S - FLEX	HANNES	7 TO 1	PAIN	PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE PASSE	STARAS	RESETA STR FRA	101-01				
CALLED BY IS USED BY CALLS	438 - ITROT - AJACOB HANU .S - HANAL	MAREN MARIA STARAD	CONSTR RINGETA STARAN	CONTRA	DER JV STAB	FOCUS	HAPPES TRIM	<b>3</b> E	1457.A4		3	10001
CALLED BY IS USED BY CALLS	600 - MING - AJACOB - STARAN	A ANAL BREKTA	CONSTD STAB	CONTRA	2	Santin	3	Busta	E		. 2	2
CALLED BY 15 USED BY CALLS CALLS USES	C70 AMAL - AJACOB RECTA S - ACOIT 1 - AMOUT TOPLOT	2 COUSTS T CLCS T COUXTS T CLCS T CLC	CONTINUED TO SAKE PORTOR	PER IV FOSIKI FORIKI	PALLES POSSER	STALAS STALAS POSUMI	STAR S	STABAS STABAS	STOWAK STRIAG	STATE TABLE	2 FF	Tablet Tablet
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Continued.	INSTAB	E S	TIME P STANA	STRIM				-	STR 104 FLEBAL	PALE
	141	STRIMA	PARKA PARKA	STARAN				r r	STR JAB INSTAR	140001
TABLE 4-1.	METRIN HPC6	STRIA	PORTY TOPLOT	START		HI THE		100-CT	STARAM	
TA	Saerre	<b>5</b>	RAIN FORTH	RTINIT		STATE		HPC6 STRIMA	STARAD	MATRIM INSTAN STRIMA
	MEINST CONTRA TRIA STRIMA	STAB MAIN STARAN STRINA	HPCG FORY STANAN	MAIN	MA IN ST&D	READIN MAIN TOPLOT	PA 100	MATATA HAPPIS STANAN	PA.TH PA.MAL PORV	STRING STRINGS
	CALLED BY — AJACOB IS USED BY — CONSTB TIME CALLS — MANAL	CALLED BY - INSTAB IS USED BY - COMSTB CALLS - MANAL USES - MANAL	LENGTH DAD CALLED BY - INIT IS USED BY - HANNIS CALLS - DATE USES - MANAL	LEMETH 10ES CALLED SV - MCAL IS USED SV - INTO CALLS - FLEX	LENGTH CFO CALLED BY - ALSTAB IS USED BY - CRYSTB CALLS - STARAN	LENGIN 1FB CALLED BY - MAIN IS USED BY - CONTRN CALLS - DATE	LENGTH 11DB CALLED BY - STAB 1S USEC BY - CONSTB CALLS - STBD	LENGTH 1230 CALLED BY - WHAMU IS USED BY - CONTRA CALLS - MANAL	LENGTH 488 CALLED BY - TAIN IS USED BY - CONTRR CALLS - DATE USES - DATE	LENGTH 640 CALLED BY - AJACOB IS USED BY - CONSID CALLS - MANAL
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4-1. Co	DERIV	10-01		Sparen			STARAB	1876 1878	
TABLE 4	CONTIN	STREM		DERIV			Talks.	VARI HANNS START	
H	CONST8 STAB	STAMA		CONTEN			FORM	TILT DERIV ENGRIA STRIMA	
	ABAL BEEKTA STRIVA	7	STRIM	COMSTO STAB STR 144			START	STAB CONTRR MTLT STAMAR	
	LENGTH 2AB CALLED BY - FUSFUM 15 USED BY - AJACOB INAUW CALLES - STARAN	LENGTH 5C0 CALLED BV - STARY IS USED BV - NAIN CALLS - INSTAR	LEMETH 200 START 15 USED BY - START 15 USED BY - MAIN CALLS - INSTAR	15 USED BY - AMA, 15 USED BY - AJACOB 16 USED BY - AJACOB 17 USED BY - AJACOB	CALLED BY - START IS USED BY - MAIN	CALLED BY - START IS USED BY - MAIN	CALLED BY - 112E IS USED BY - MAIN CALLS - FLEX	15 USED BY - AJACOB 15 USED BY - CONSTB 16 USED BY - CONSTB 17 USED BY - CONSTB 18 USED BY - CONSTB	
	33# 30#	NOME TO SERVICE AND ADDRESS OF THE SERVICE AND A	THE STATES	NSTORE N		100	2002	זונא	

TABLE 4-2. CRCSS-REFERENCE OF MULTIPLE ENTRY SUBROUTINES IN AGAJ73

Name of Additional Entry Point	Contained in Subroutine	Name of Additional Entry Point	Contained in Subroutine
AJACB1	AJACOB	RESTO	RESTRT
ALLVEC	ALLMAT	REST1	RESTRT
ATMINT	JFBGIN	REST2	RESTRT
ATRM	TRIM	REST4	RESTRT
AUXJ	CNTM	RWKOUT	REDRWK
AZMOUT	RADOUT	SUBA	MOMB
CGXARM	CGZARM	SUBB	MOMB
CGYARM	CGZARM	SWKOUT	REDSWK
FLAP	CNTM	TFFA	TILT
FLAT	MTLT	TTLT	MTLT
HARM1	HARM	vcnt1	VCNTRL
HSAF	TILT	VCNT2	VCNTRL
INBLDl	INBLD	VCNT3	VCNTRL
INBLD2	INBLD	VCNT4	VCNTRL
INBLD3	INBLD	VCNT5	VCNTRL
INBLD4	INBLD	VCNT6	VCNTRL
IOWRFM	WRINST	VCNT7	VCNTRL
IOWRF1	WRINST	VCNT8	VCNTRL
NOPS1	NOPS	VCNT9	VCNTRL
nsted1	nsted	VCNT10	VCNTRL
PDZ	PDZ1	WRMDL1	WRMDAL
RATI	DAMPER	WRMDL2	WRMDAL
RESTOR	SAVEIC	WROT	WROT1
RESTR1 .	SAVEIC	YAWP	MTLT

GDAJ07	
POR	
TABLE 4-3. CONTROL SECTION CROSS-REFERENCE FOR GDAJ07	
SECTION	
CONTROL	
TABLE 4-3.	

0000	-	1	ï	Chicara
POX C	TY COLUMN	, VI	3	KERDARD
PURG	Inchibr		NOILO	REMCVED
PURG	IHCCOMMS	CONTROL SF	CTICN	REMOVED
PURG	IHCERRM		CTION	REMOVED
PURG	THCLSORT	TROL	CT ION	REMOVED
PURG	THCFCVTH	IRDL	CTION	REMOVED
PURG	IMCSSORT		CTION	REMOVED
PURG	IHCEFIOS		CTION	REMOVED
Dand	IHC UAT BL		SECTION	REMOVED
PURG	IHCEFNTH		ECT I CA	REMOVED
PURG	IHCSSCM		CTION	REMOVED
PURG	INCETACH	CONTROL SE	CTICN	REMOVED
PURG	IHCLÁTNZ		CTION	REMUVED
PURG	IHCSLOG		CTION	REMOVED
PURG	IHCSEXP		ECT ION	REMOVED
PURG	IHCFRXPR		ECT ION	REMOVED
PURG	IHCSATNZ		ECTION	REMOVED
PURG	INCIBERH		ECTION	REMUVED
PUNG	IHCFRXPI		CTION	REMOVED
PHRE	NOS TONI		SCTION	REMOVED

				TA	TABLE 4-3.	- 1	Continued.					
10041448	LENGTH 30											
SPRI V002	LENGTH 60											
AKI Se	LENGTH 8D0 CALLED BY 1S USED BY CALLS USES	D PLOTER - CALCON	FSFT MUNDER NEXTTINE	NAIN SYMBOL SYMBOL								
CACON	CALLED BY	FO AXISS	1831V3 CVIC81	COMPLT	LINE	HAIN LINE	PLOTER MAIN	SCAL IT	SYMBOL PLOTER	SCALIT	•	
CALCSI	LENGTH 1498 CALLED NY - 1S USED -BY - CALLS - USES -	- 5000LT - 5000LT - 5450M - 5450M	MAIN HEDING NEXTINE	INPLOT	LINE	NUMBER	SYMBOL	TOPLOT				
1700	CALLED RY CALLED RY CALLS USES	- CACON - AXISE PLOTDI	CURVET CALCON PLOTER	CALCOIL CALCOIL PPLOT	FSFT DATE . SCALED	SCAL IT HARN SCLFIX	Tr~ OT HEDING SYNBOL	INPLOT	LINE	MAXMIN WROT1	MEXTTIME NUMBER	7. OTO
CUAVET	LEMCTH: 1090 CALLED BY IS USED BY CALLS USES	90 - COHPL T - MAIN - HEDING - DATE	T18675	T0PL0T	100L01							-
C811	CALLED BY IS USED BY CALLS	- COMPLT - HAIN - MAKAIN	TIMPTS	100.01								
DATE	LENGTH 1FO CALLED BY	F0 - WROT1 - CONPLT	CURVET	MAIM	PP.07	SCAL IT				•		-
53	CALLED BY IS USED BY CALLS USES	COMPLT - MAIN - MARN - AXIS	HEDING CALCON	PLOTER LINE	TOPLOT NEXTTINE NUMBER	NUMBER	2,010	PLOTO1	SCALES	SYMBOL		<del></del>

O

15   15   15   15   15   15   15   15					TABLE	E 4-3.	Continued	ned.						
Substitute of the continue   Co	11761	<del> </del>	FSFT COMPLT					,						
15 USEO N° - COMPLY   MAIN   SCALIT SALTIN	F0146	• •	CALCOI COMPLT	CURVET HAIN PLOTO1	FSFT SCALIT	PP.01		•						
CALLED WY - CALCEN   FSPT   MAIN   SCALIT		CALLED BY -	101	_ 9.5		NOL SECTI SCLFIX	8							
CALLE OF A LAISE CACON COMPLY NATHIN TOPLOT WIND!   CALLE OF ALTS   CACON CA		LFMCTH *** CALLED BY IS USED BY CALLS	CALCON CONPLT CALCON	PLOTER FSFT SYMOL NEXTINE		SCAL IT								
CALLED BY - CONPLY   ALIN SCALIT   CALLED BY - CONPLY   CALLED BY - CONPLY   ALIN SCALIT   CALLED BY - SYMBOL   CALCED BY - AXISS   CALCED BY - AXISS   CALCED BY - AXISS   CALCED BY - CONPLY   FSFT   LINE   NAIN   NUMBER   PLOTER   CALLED BY - CONPLY   FSFT   NAIN   PPLOT   SCALIT   CALLED BY - CALCED BY - CALCED   CONPLY   CURVET   FSFT   NAIN   PPLOT   SCALIT   CALLED BY - CALCED BY - CALCED BY - CALCED BY - CALCED   CONPLY   CURVET   FSFT   NAIN   PPLOT   SCALIT   CALLED BY - FSFT   STATSS   CALCED   SYMBOL   STATSS   STAT		LENGTH 1436 CALLS - USES -	CALCON AXISO MEXTTIME	COMPLT CALCON	MAKMIN CALCOI PLOTO	TOPLOT CLAVET PLOTO1	WROTI Cell PLOTER	DATE	FSFT SCALED	HARN SCALIT	HEDING SCLFIX	INPLOT	LINE	MAXPIN
S USED BY - SYMBOL   S   S   S   S   S   S   S   S   S	MAKINTA		THIS CONPLT		SCLFIX SCALIT	MDL SECTI	5							
LEWCTH   170	NEXTTINE			CALCOL	CONPLT	FSFT	¥	MAIN		PLOTER	SCAL IT			
LENGTH 1A48 CALLED BV - HEDING IS USED BV - CALCEL COMPLT CURVET FSFT NAIN PPLOT LENGTH AAD CALLED BV - HEDING IS USED BV - CALCEL COMPLT CURVET FSFT NAIN PPLOT LENGTH 478 CALLED BV - FSFT IS USED BV - FSFT IS USED BV - CALCON LINE NUMBER SCALES SYNBOL USES - CALCON REXTINE NUMBER SYNBOL	#140F#	LEMCTH IT	AXISE COMPLT SYMBOL CALCOR	CALCBI FSFT NEXTTINE		PLOTER	SCALIT				•			
LENGTH AAD CALLED BY - HEDING IS USED BY - CALCAI COMPLT CURVET FSFT MAIN PPLOT LENGTH 478 CALLED BY - FSFT IS USED NY - CONPLT MAIN IS USED NY - CONPLT MAIN USES - CALCON MEXTTIME MUNDER SYMBOL	201	₹ .		COMPLT	CURVET	FSFT	MAIM	10744	SCALIT					•.
LENGTH 478 CALLED BY - FSFT IS USED BY - COMPLT HAIN IS USED BY - COMPLT HAIN CALLS - AXIS# CALCON LINE HUMBER SCALES USES - CALCON NEXTTINE MUMBER SYMBOL	A GTO1	3	HEDING CALCOL	COMPLT	CURVET	FSFT	# T T	PPLOT	SCALIT					
	PLOTER.	4 4	FSFT COMPLT AXISA CALCON	HAIN CALCON NEXTTINE	CINE	NUMBER	SCALES	SYMBOL						

			7	TABLE 4-3.		Concluded						
10.4	LENGTH 780 CALLED BY - SCALIT IS USED BY - COMPLT CALLS - MEDING USES - DATE	17 RAIN 16 INPLOT	MEGT1	100.01	1							
SCAL E.	CALLED BY - PLOTER IS USED BY - COMPLT	54 17 1961	MIM									
20417	LENGTH   19740   CALLED BY - CALCON   CALLS - CALCON   CALLS - CALCON   USES - USE	CALCEL	1891.07 HED 186	PPLOT	SCLFTX	TOP, OF HAZHEN	HEXTTINE MUNDER	MUNDER	P. 070	P.Of01	SYMBOL	10-01
SCLFIX	LENGTM 438 CALLED BY - SCALIT IS USED BY - COMPLY CALLS - INPLOT	IT MAIN T HAXMIN	101-01									
SYMBOL	LEMETH BCS CALLED BY - AXISS IS USED BY - AXISS CALLS - CALCON		LINE COMPLT	MUNDER FSFT	PLOTER MAIN	PLOTER	SCAL IT					
T 4 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1	LFWGTH 2C35 THIS CALLED BY - CURVET IS USED BY - COMPLT	IS IS A "CON	. COL	IS A "COMMON" CONTROL SECTION COLL MAIN	8							
10-01	CALLED BY - CALCEL IS USED BY - COMPLY	_	CURVET MAIN	S A "COMMON" CONTROL SECTION COMPLT CURVET CALL F. CURVET MAIN PPLOT SA	PSFT SCALIT	#14#	SCAL17	SCLFIX	MOT 1			
MOT1	LENGTH 200 CALLED BY - CURVET IS USED BY - COMPLY CALLS - DATE	ET MAIN LT MAIN TOPLOT	PPLOT									
												-

		TABLE	TABLE 4-4. REDUCTION	REDUCTION OF AGAJ73 STORAGE REQUIREMENTS	TORAGE REQUI	REMENTS		
Array	Array Name	COMPON	ĬQ	Dimensions of	Array		Approx. Reduction of	ion of
Outside	In RESTRT	Name	Outside RESTRT		In RESTRT	TRT	Storage Requis.**	ts. ##
RESTRT			Standard	Reduced	Standard	Reduced	$Bytes(x10^{-3})$	X
WRRTR	FRK1	FORWK1	(33,10,5,20,2) (1,1,1,1,1)	(1,1,1,1)	(00099)	(1)	797	260
WKSTB	FSK1	POSWK1	(15,10,5,12)	(1,1,1,1)	(0006)	(1)	36	*
CURVED	TABL	TAB	(1100,5)*	(1100,2)*	(6350)	(3050)	13	12
CURVEL	TABL1	TABI	(500,5)*	(500,2)*	(5375)	(2150)	9	9
CURVEA			(575,5)*	(575,2)*		),,,,,,	7	9
AUX	FYY1	FORTY1	(16,215)	(1,1)	(3440)	(1)	14	12
*In bloc reduced block of NACA OF TABOUT subrout	*In block data DAT1, the reduced array is 1 rather block data (e.g., CL5A, NACA 0012 airfoil. Also TABOUT must be changed. subroutine TABFIX in TAB	I, the sect l rather th CL5A, CL5E Also, if langed. The in TABOUT	In block data DATI, the second subscript of the standard array is 4 rather than 5 and that of the reduced array is 1 rather than 2; the remaining entries of the COMMON block are filled with the block data (e.g., CL5A, CL5B, etc., for TAB). The block data in DATI are the data tables for an NACA 0012 airfoil. Also, if TAB and TABI are reduced, three statements near the end of subroutine TABOUT must be changed. The location of each statement is given with respect to the last call to subroutine TABFIX in TABOUT (i.e., the nth statement after the call).	the standard ar ining entries of . The block dat ire reduced, thre ich statement is statement after	array is 4 of the COMMO lata in DATI ree stateme is given wit	rather than N block are are the da nts near th h respect t	filled with the fitaled with the tables for the end of subroto the last cal	the he an utine 1 to
	Location		Standard			Reduced	83	
	1st 5th	H &	IP (K.EQ.5) RETURN K=5	5	IF (K.Eq	IF (K.EQ.2) RETURN W=?		
	7 <sup>th</sup>		IF (YRR(18,1).Eq.5OR. YSTBZ(18,1).Eq.5.) GO TO 26	5OR. ) GO TO 26	IF (YRR) YSTBZ(18	IF (YRR(18,1).Eq.2OR. YSTBZ(18,1).Eq.2) GO TO 26	. OR. 30 TO 26	
**IK = 1024 bytes	124 bytes							

TABLE 4-5. STAB	DIAGNOSTIC SWITCH IN AGAJ73
IPL(34)	Variable
1	FUS. U
2	FUS. W
3	FUS. Q
4	FUS. V
5	FUS. P
6	FUS. R
7	M.R. F/A PYLON RATE
. 8	M.R. LAT PYLON RATE
9	T.R. F/A PYLON RATE
10	T.R. LAT PYLON RATE
11	M.R. F/A FLAP. RATE
12	M.R. LAT FLAP. RATE
13	T.R. F/A FLAP. RATE
14	T.R. LAT FLAP. RATE
15	M.R. F/A PYLON DISP
16	M.R. LAT PYLON DISP
17	T.R. F/A PYLON DISP
18	T.R. LAT PYLON DISP
.19	M.R. F/A FLAP. DISP
20	M.R. LAT FLAP. DISP
21	T.R. F/A FLAP. DISP
22	T.R. LAT FLAP. DISP

Figure 4-1. Job Control Language to Run AS812A.